

DISTANCE TO CLEAR 50 FOOT OBSTACLE - WATER TAKEOFF

TWO ENGINES

**MODEL
CH/HH-3E**

**ENGINE
T58-GE-5**

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

**CONDITIONS:
FOD SHIELD ON OR OFF
MAXIMUM POWER
TECHNIQUE: LEVEL
ACCELERATION
FROM WATER
103% N_r**

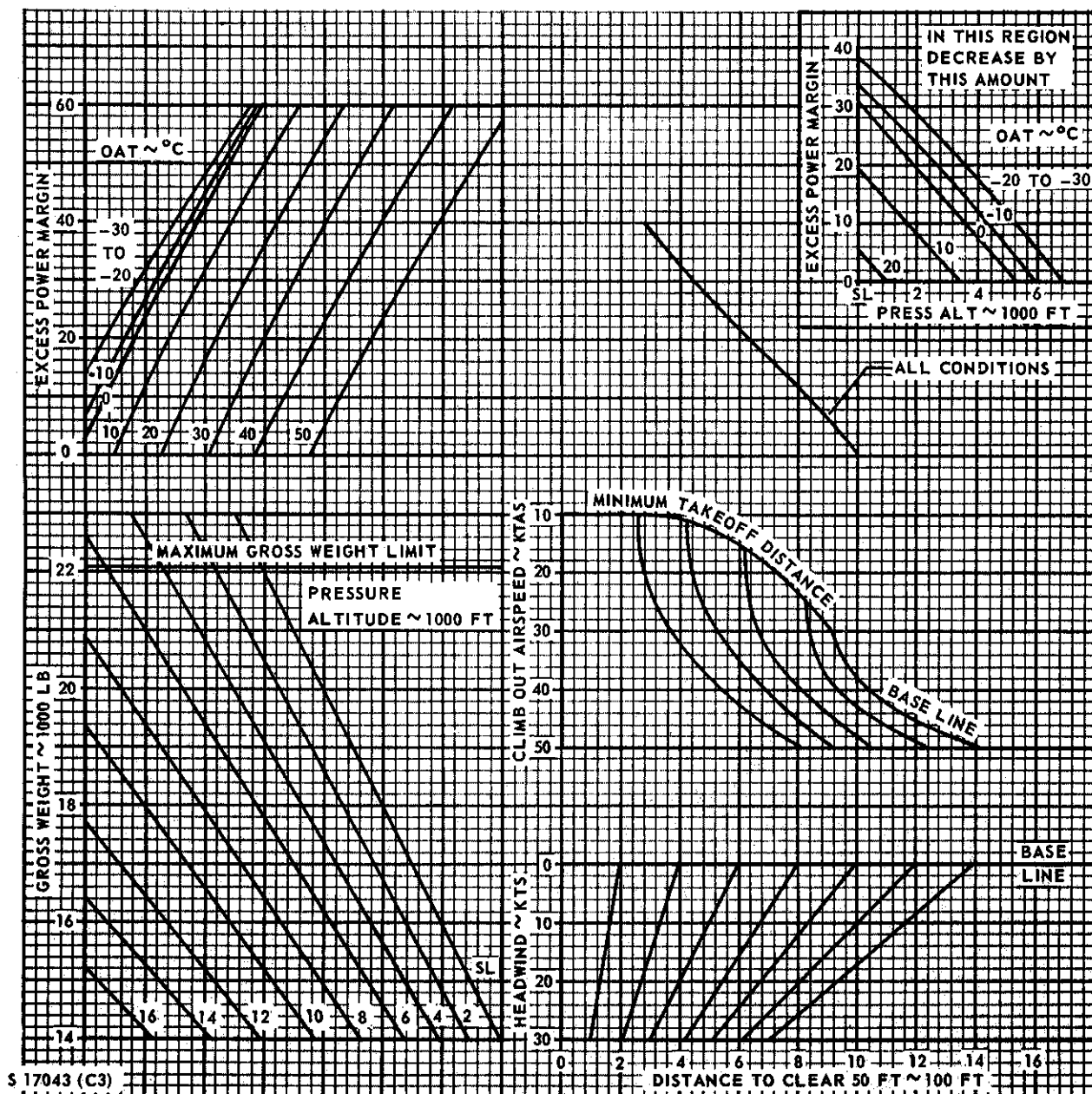
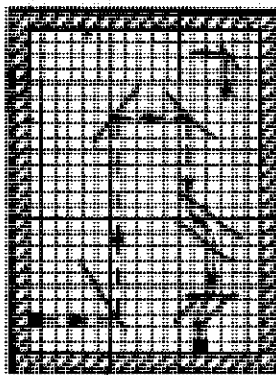
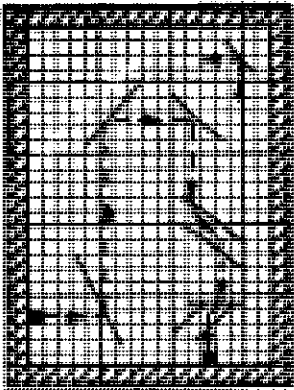


Figure A-17. Distance to Clear 50-Foot Obstacle - Water Takeoff - Two Engines

CONDITIONS:
FOD SHIELD ON OR OFF
MAXIMUM POWER
TECHNIQUE: ROLLING
103% N,



**DISTANCE TO CLEAR 50
FOOT OBSTACLE -
ROLLING TAKEOFF**

TWO ENGINES

MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 15 APRIL 1971
DATA BASIS: FLIGHT TEST (AIR FORCE)

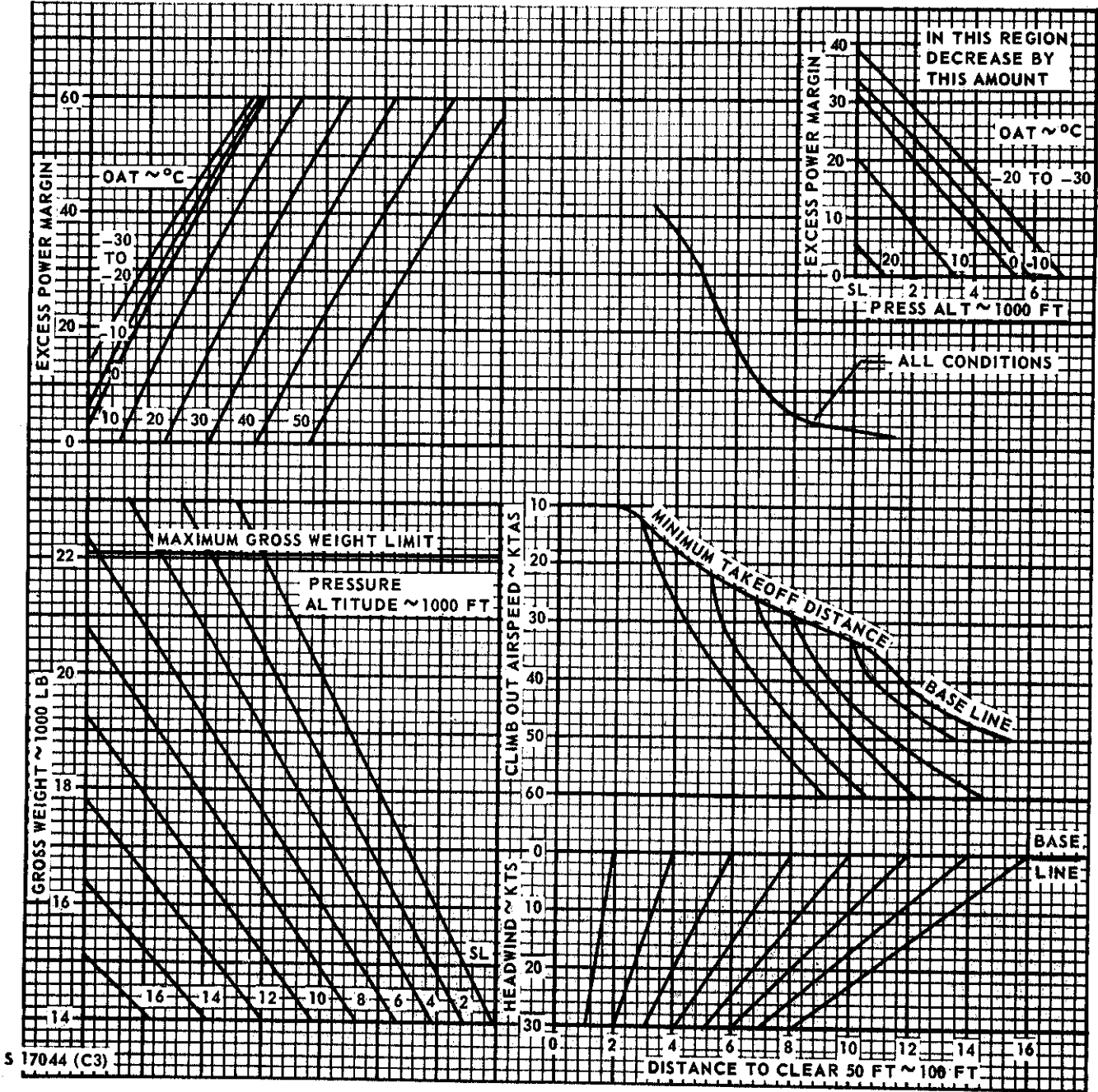


Figure A-18. Distance to Clear 50-Foot Obstacle - Rolling Takeoff - Two Engines

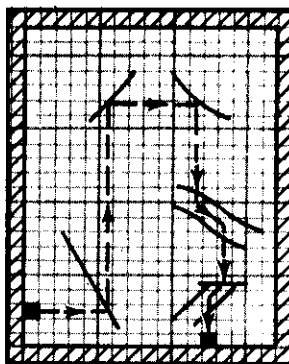
DISTANCE TO CLEAR 50 FOOT OBSTACLE - ROLLING TAKEOFF

ONE ENGINE

MODEL
CH/HH-3EENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



CONDITIONS:
FOD SHIELD ON OR OFF
MAXIMUM POWER
TECHNIQUE: ROLLING
103% N_T

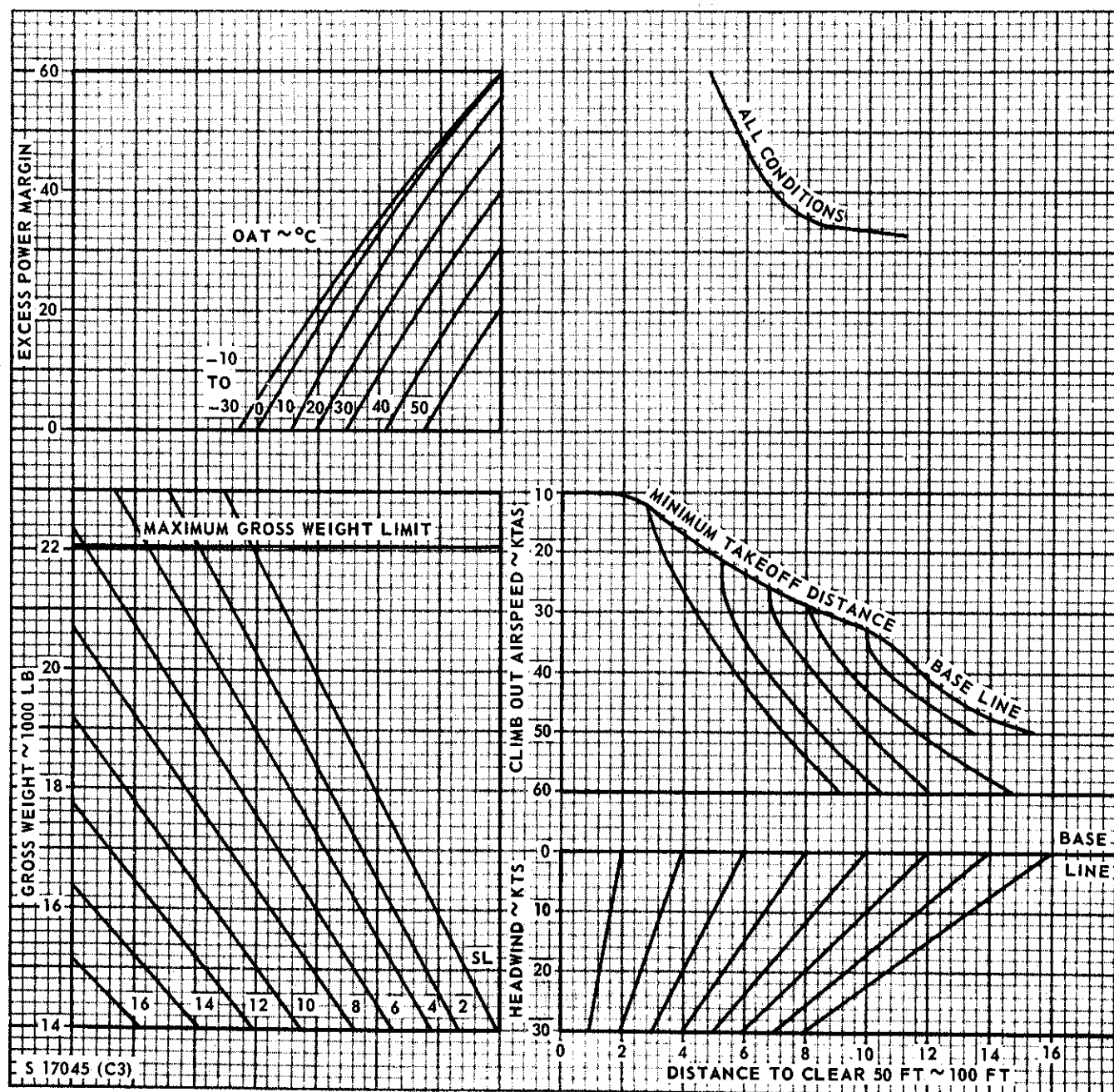
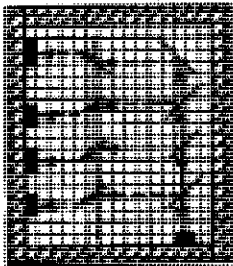


Figure A-19. Distance to Clear 50-Foot Obstacle - Rolling Takeoff - One Engine

CLIMB
TWO ENGINES
WARM DAY

MODEL
CH/HH-3E

ENGINE
T58-GE-5



CONDITIONS
103% N₁
MILITARY POWER
WARM UP AND TAKE OFF
FUEL NOT INCLUDED
ZERO WIND
FOD SHIELD ON OR OFF

DATE: 15 APRIL 1971
DATA BASIS: FLIGHT TEST (AIR FORCE)

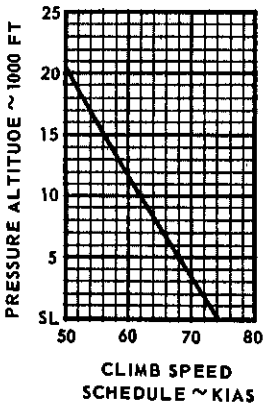
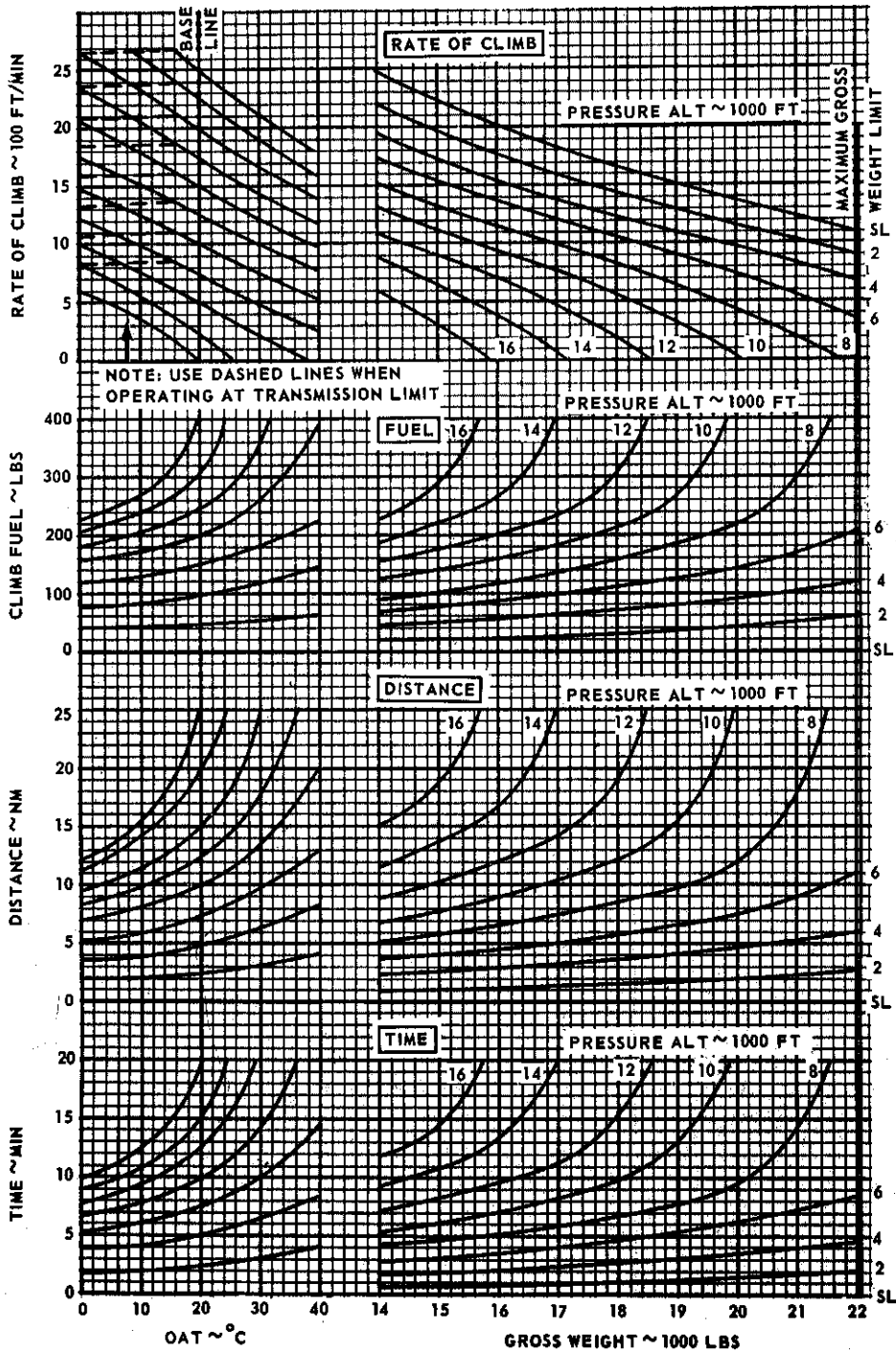
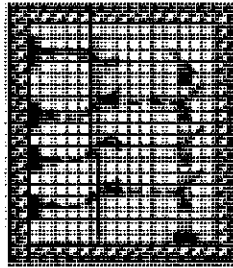


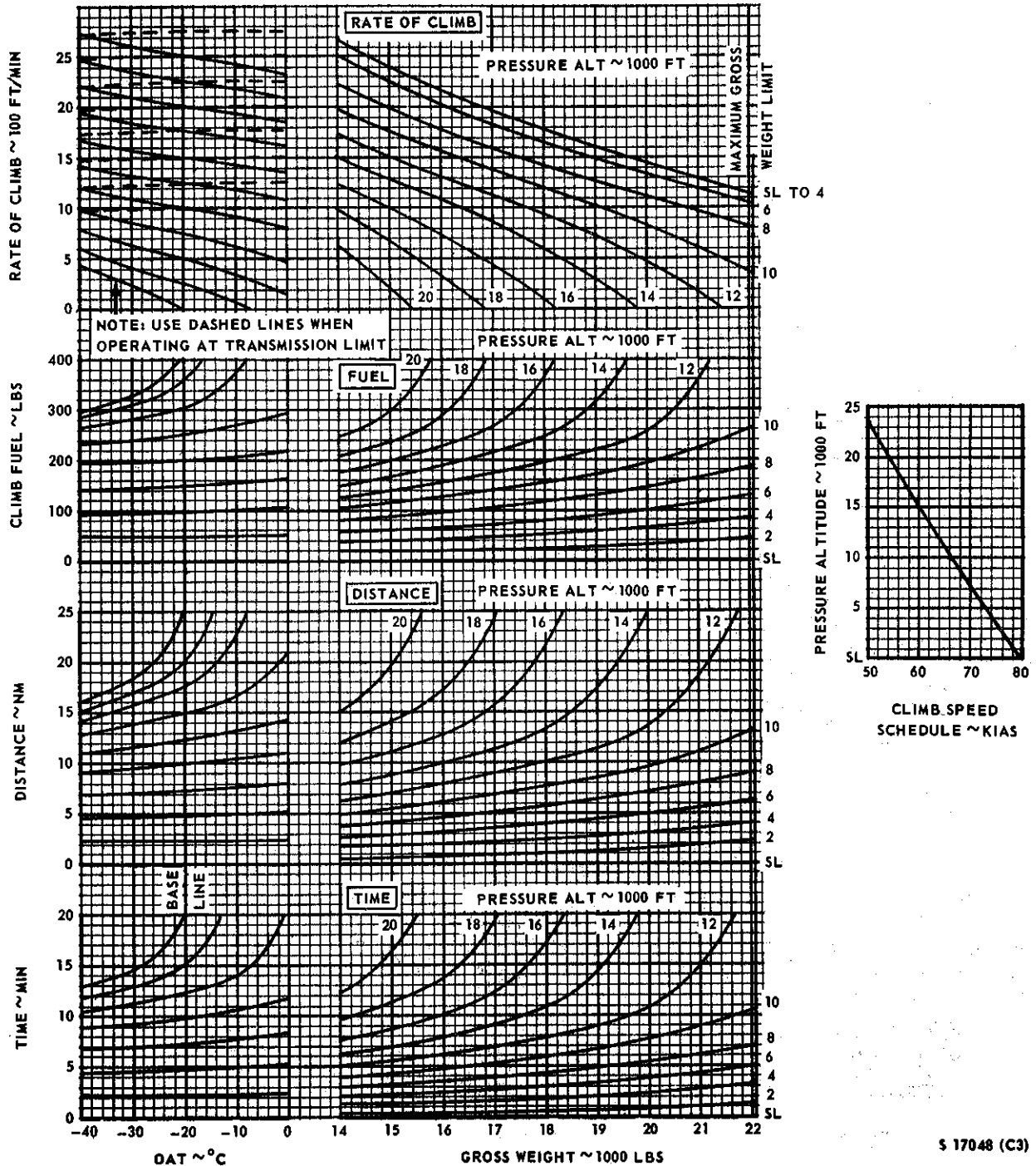
Figure A-20. Climb - Military Power (0°C to 40°C OAT) - Two Engines

CONDITIONS:
 103% N₁
 MILITARY POWER
 WARM UP AND TAKE OFF
 FUEL NOT INCLUDED
 ZERO WIND
 FOD SHIELD ON OR OFF



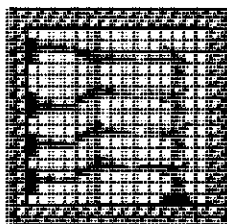
CLIMB
 TWO ENGINES
 COLD DAY
 MODEL CH/HH-3E ENGINE T58-GE-5

DATE: 15 APRIL 1971
 DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17048 (C3)

Figure A-21. Climb - Military Power (-40°C to 0°C OAT) - Two Engines

CLIMB**ONE ENGINE
WARM DAY****MODEL
CH/HH-3E****ENGINE
T58-GE-5**

CONDITIONS:
 103% N_T
 MILITARY POWER
 WARM UP AND TAKE OFF
 FUEL NOT INCLUDED
 ZERO WIND
 FOD SHIELD ON OR OFF

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

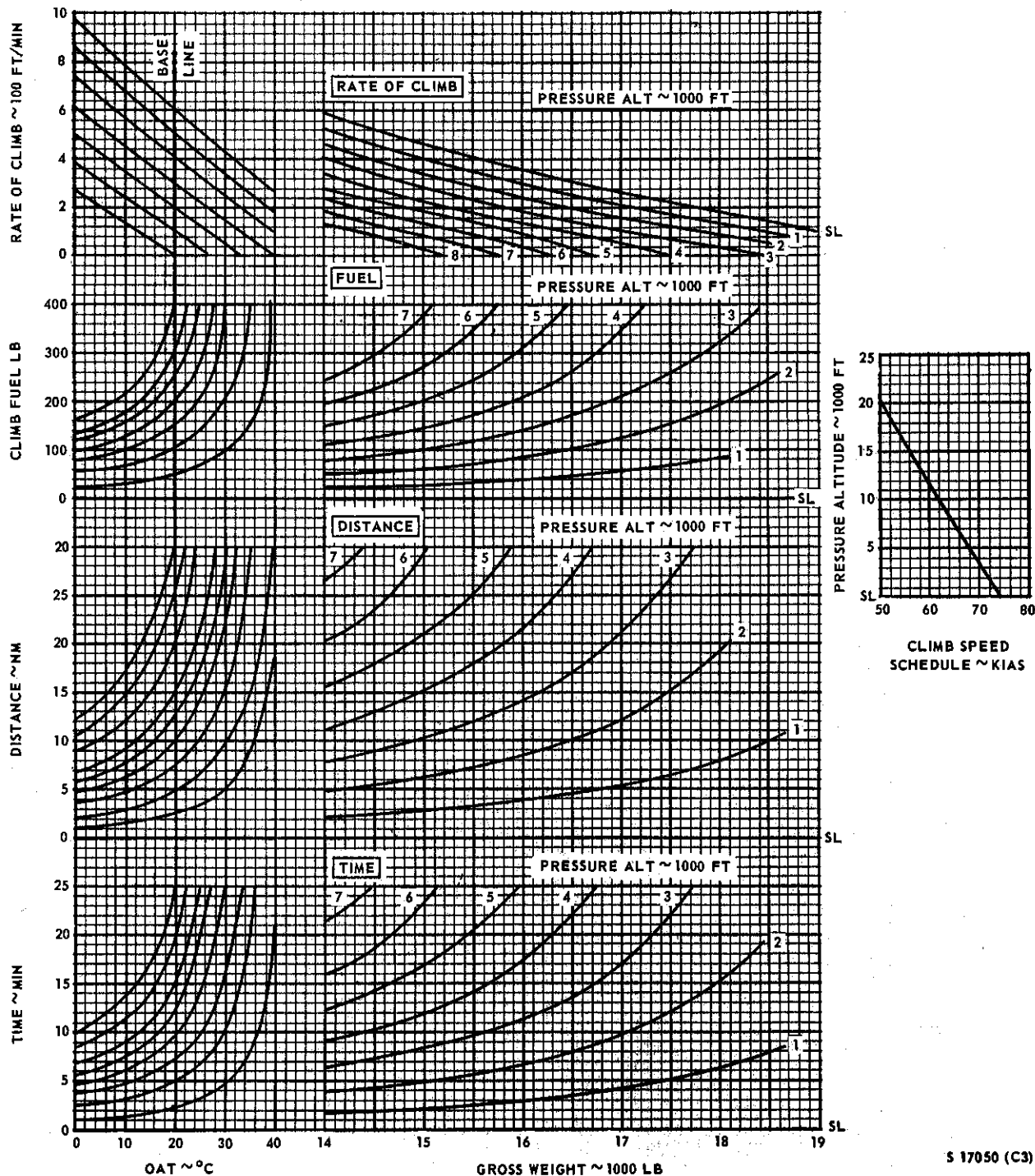
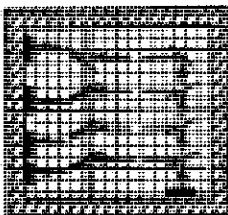


Figure A-22. Climb - Military Power (0°C to 40°C OAT) - One Engine

CONDITIONS:
 103% N₂
 MILITARY POWER
 WARM UP AND TAKE OFF
 FUEL NOT INCLUDED
 ZERO WIND
 FOD SHIELD ON OR OFF



CLIMB

ONE ENGINE
COLD DAY

MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

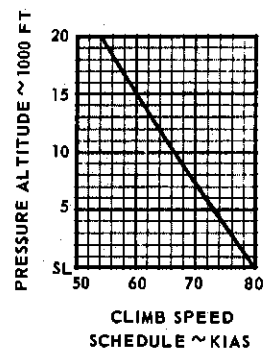
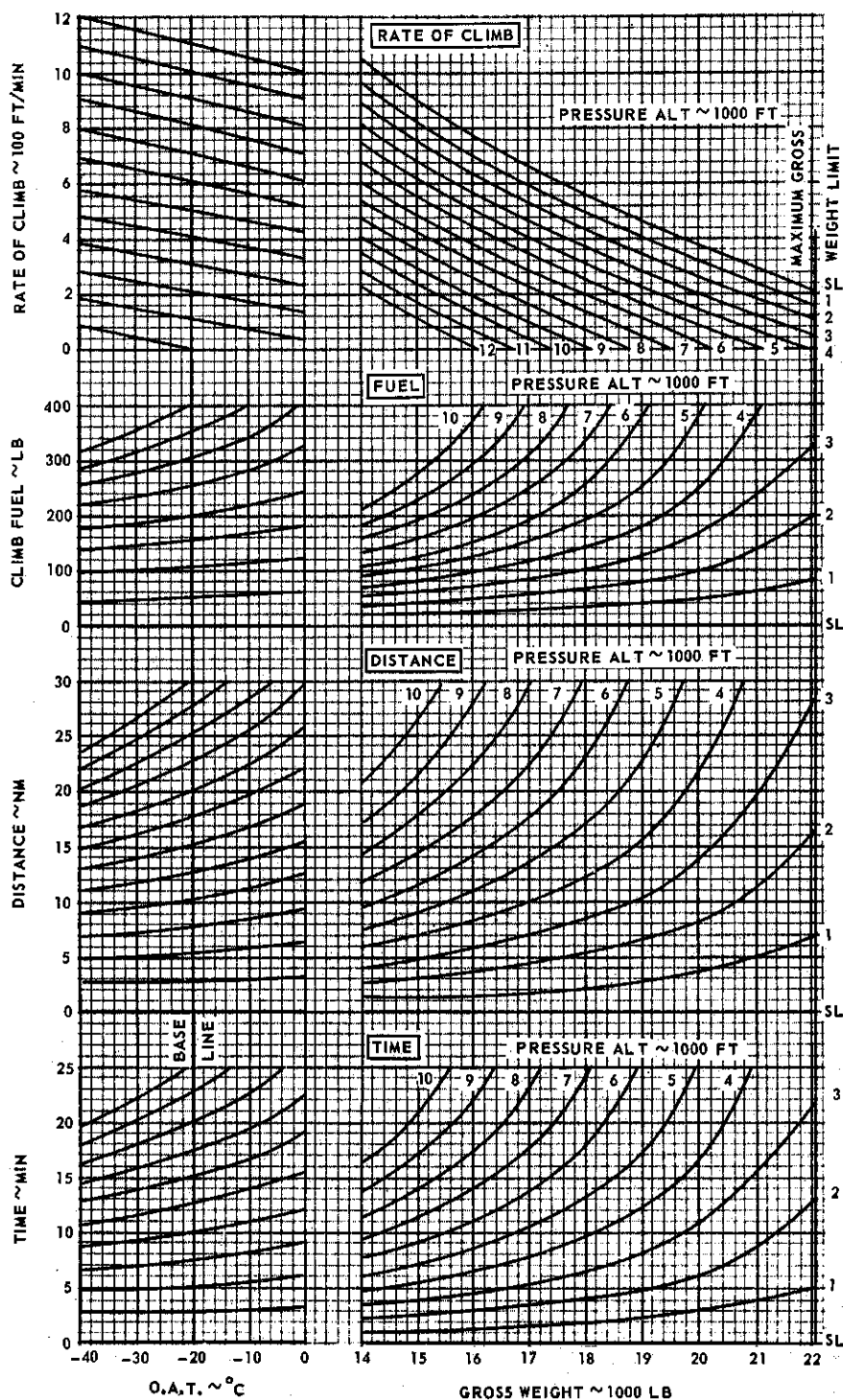


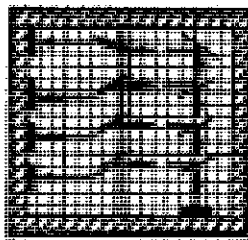
Figure A-23. Climb - Military Power (-40°C to 0°C OAT) - One Engine

S 17051 (C3)

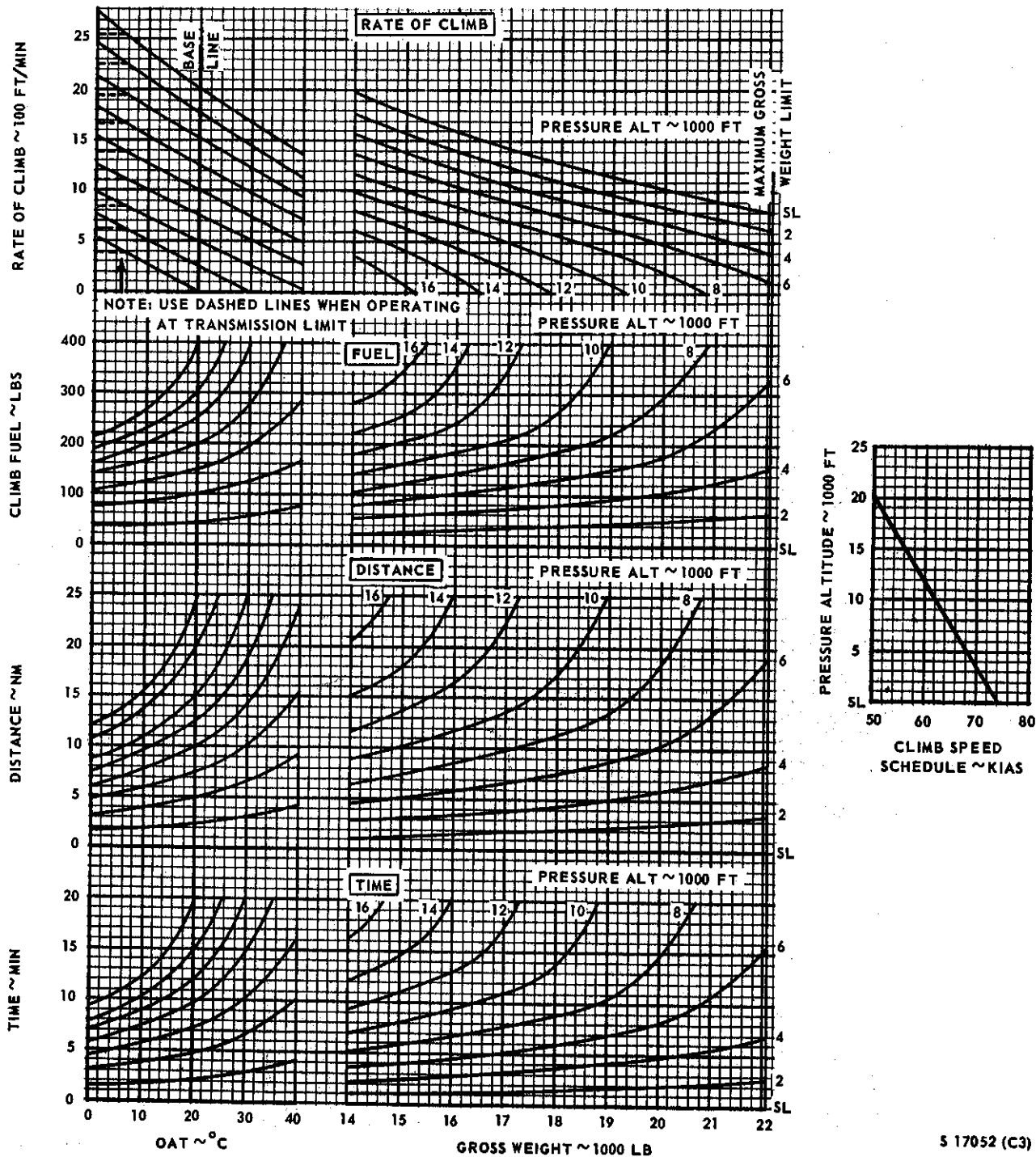
CLIMBTWO ENGINES
WARM DAYMODEL
CH/HH-3EENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

**CONDITIONS:**

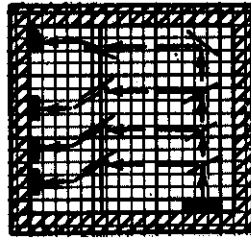
103% N_1
 MAXIMUM CONTINUOUS POWER
 WARM UP AND TAKE OFF
 FUEL NOT INCLUDED
 ZERO WIND
 FOD SHIELD ON OR OFF
 TRANSMISSION LIMIT IS
 30 MIN AT 103% Q



S 17052 (C3)

Figure A-24. Climb - Maximum Continuous Power (0°C to 40°C OAT) - Two Engines

CONDITIONS
 103% N₁
 MAXIMUM CONTINUOUS POWER
 WARM UP AND TAKE OFF
 FUEL NOT INCLUDED
 ZERO WIND
 FOD SHIELD ON OR OFF
 TRANSMISSION LIMIT IS
 30 MIN AT 103% O

**CLIMB**

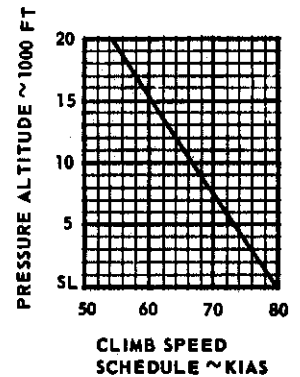
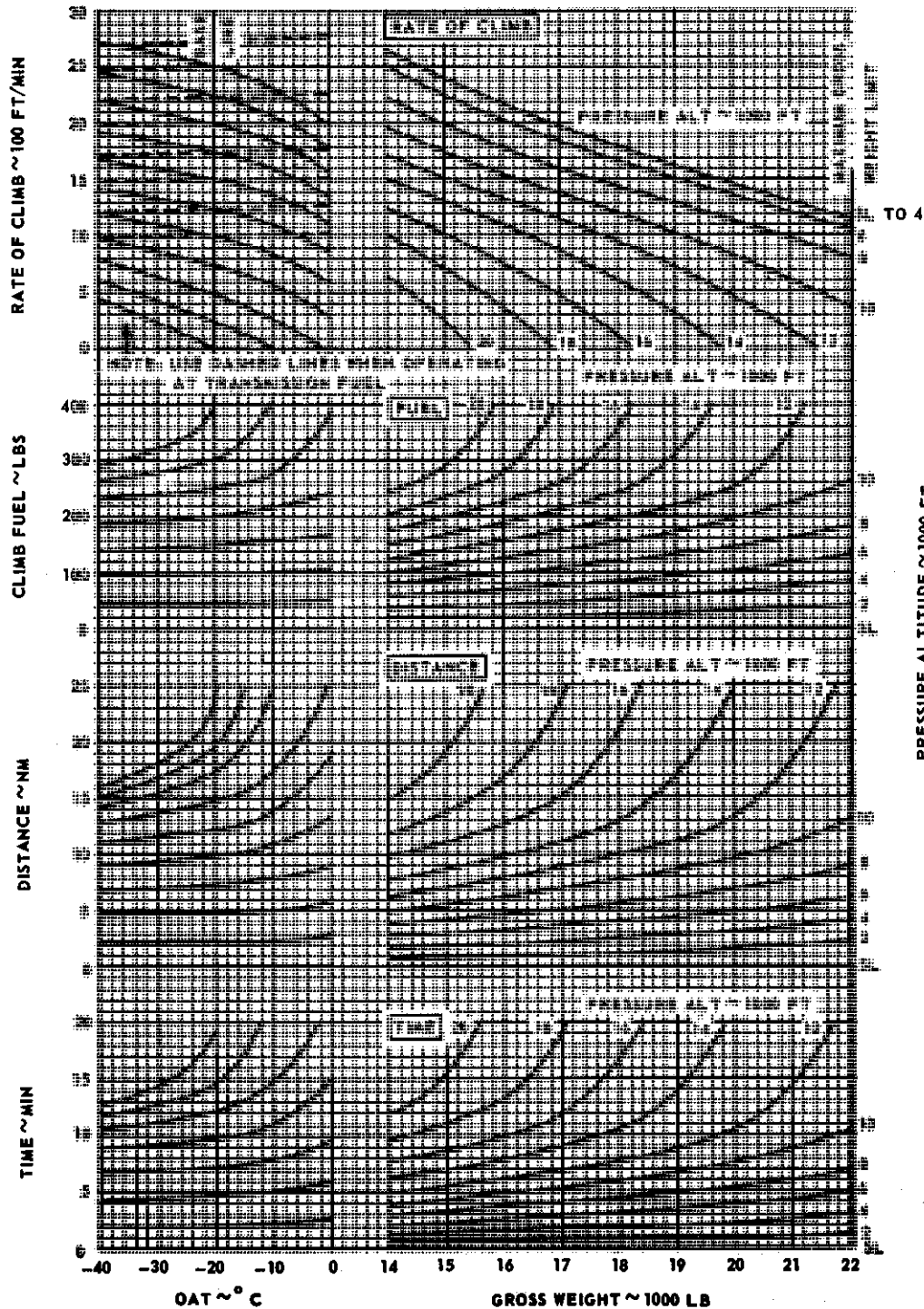
TWO ENGINES
 COLD DAY

MODEL
 CH/HH-3E

ENGINE
 T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17053 (C3)

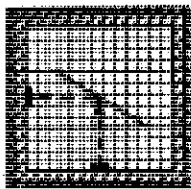
Figure A-25. Climb - Maximum Continuous Power (-40°C to 0°OAT) - Two Engines

SERVICE CEILING

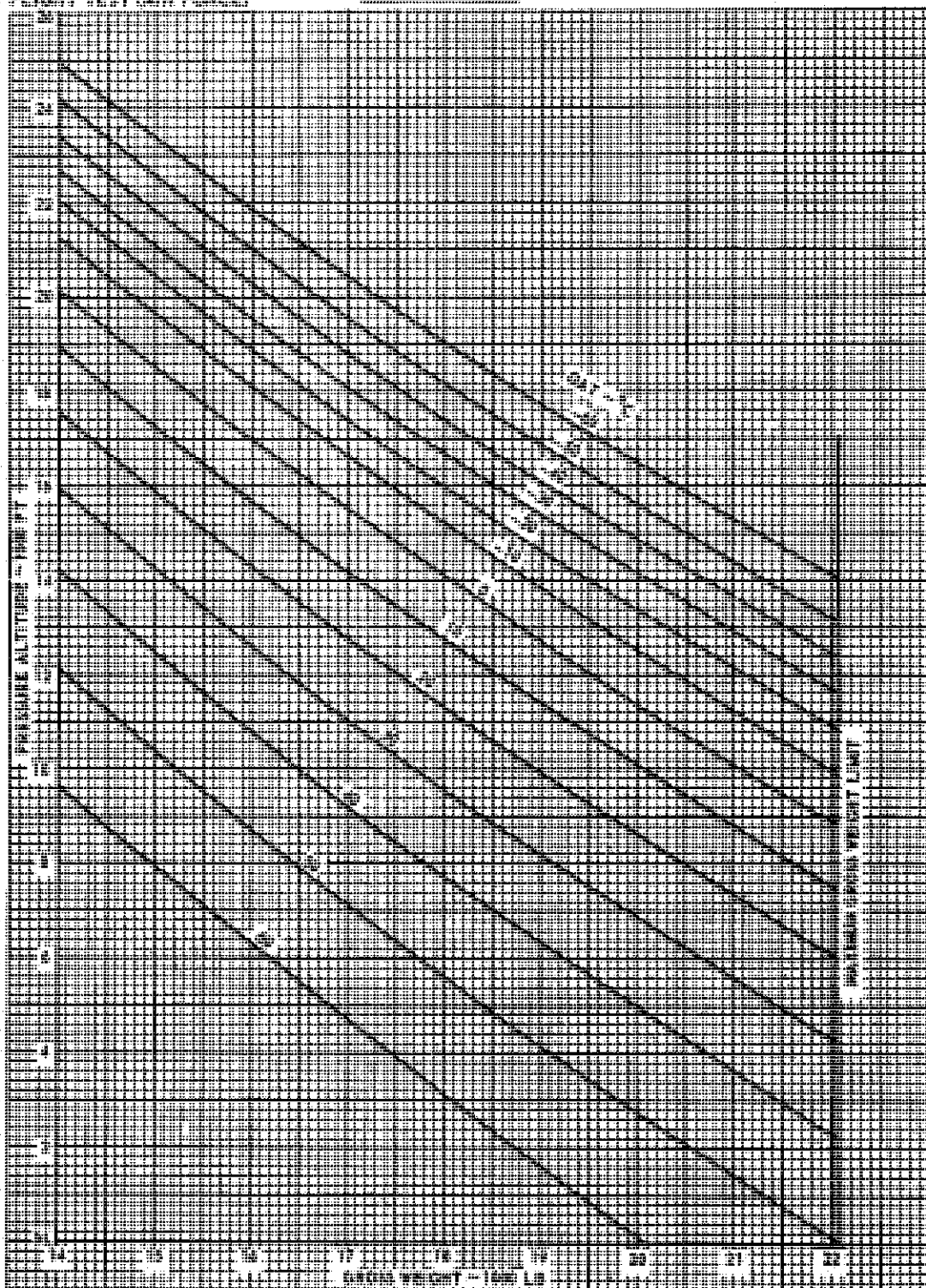
MODEL TWO ENGINES ENGINE
CH/HH-3E T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



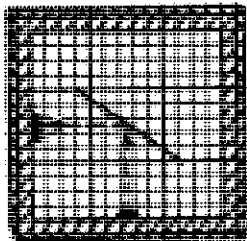
CONDITIONS:
MAXIMUM CONTINUOUS POWER
103% N_r
BEST CLIMB SPEED
FOD SHIELD ON OR OFF
100 FT/MIN RATE OF CLIMB



S17054(C3)

Figure A-26. Service Ceiling - Maximum Continuous Power - Two Engines

CONDITIONS:
 MILITARY POWER
 103% N_r
 BEST CLIMB SPEED
 FOD SHIELD ON OR OFF
 100 FT/MIN RATE OF CLIMB



SERVICE CEILING

ONE ENGINE

MODEL
 CH/HH-3E

ENGINE
 T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

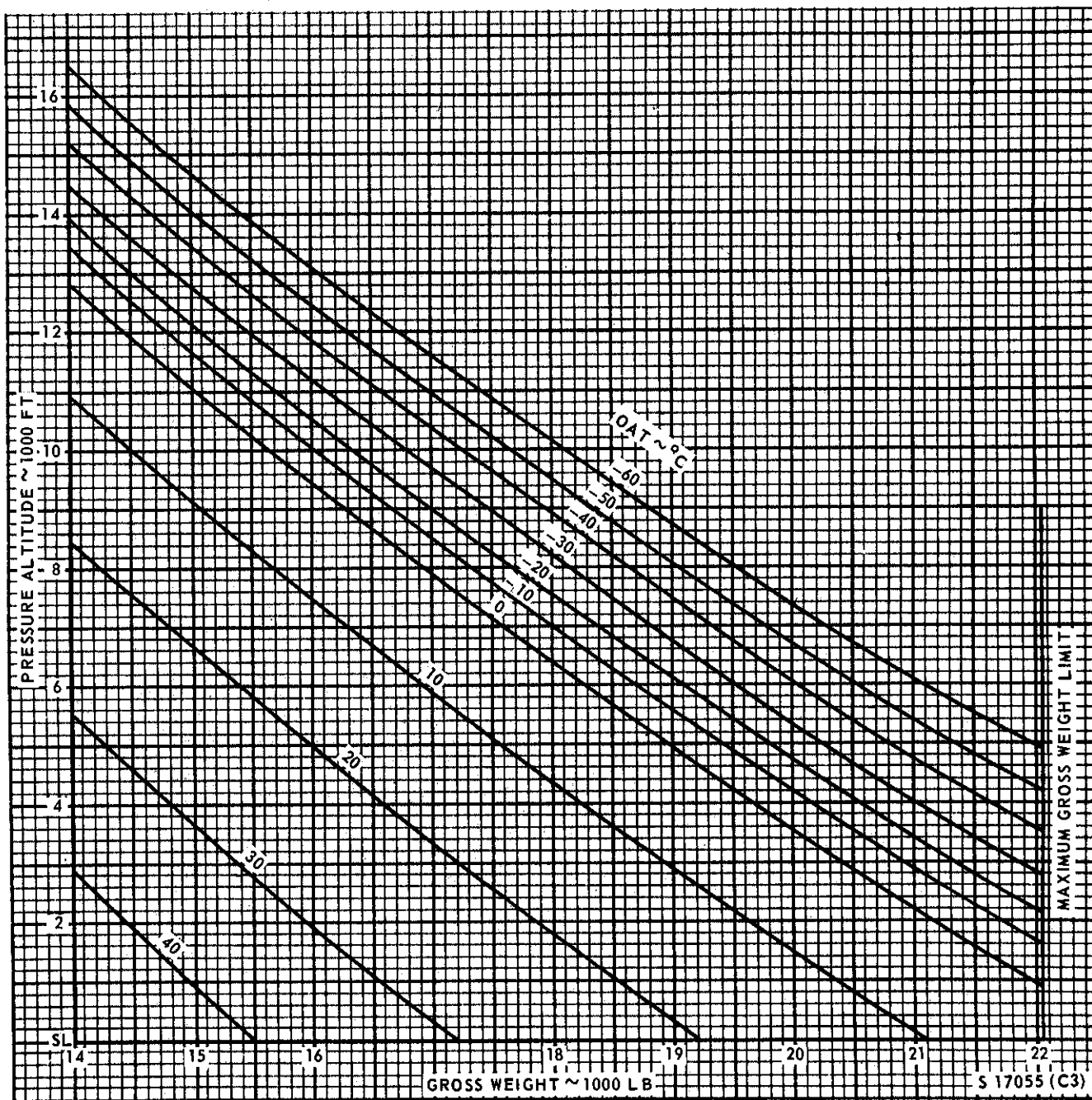


Figure A-27. Service Ceiling - Military Power - One Engine

CRUISE

OAT BETWEEN 20°C AND 40°C
TWO ENGINES

MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 2 NOVEMBER 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

CONDITIONS:

103% N_T

ZERO WIND

FOD SHIELD ON OR OFF

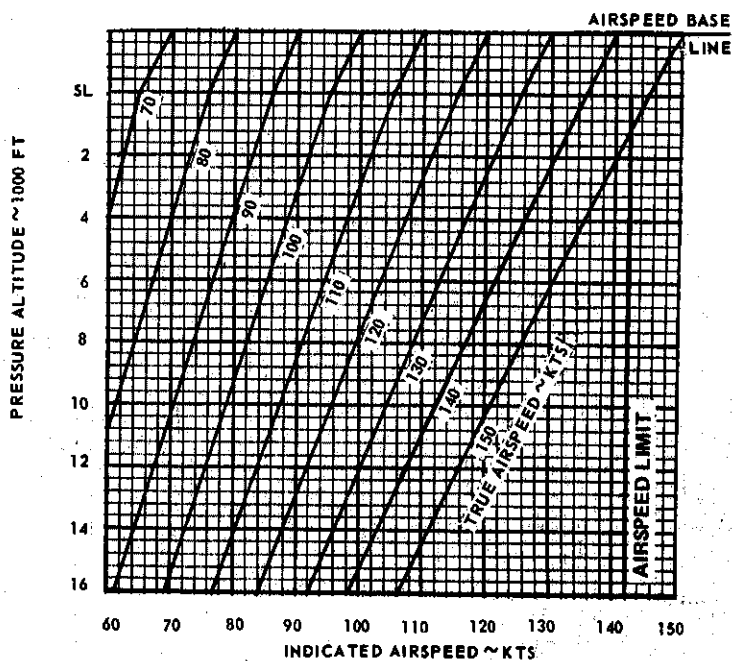
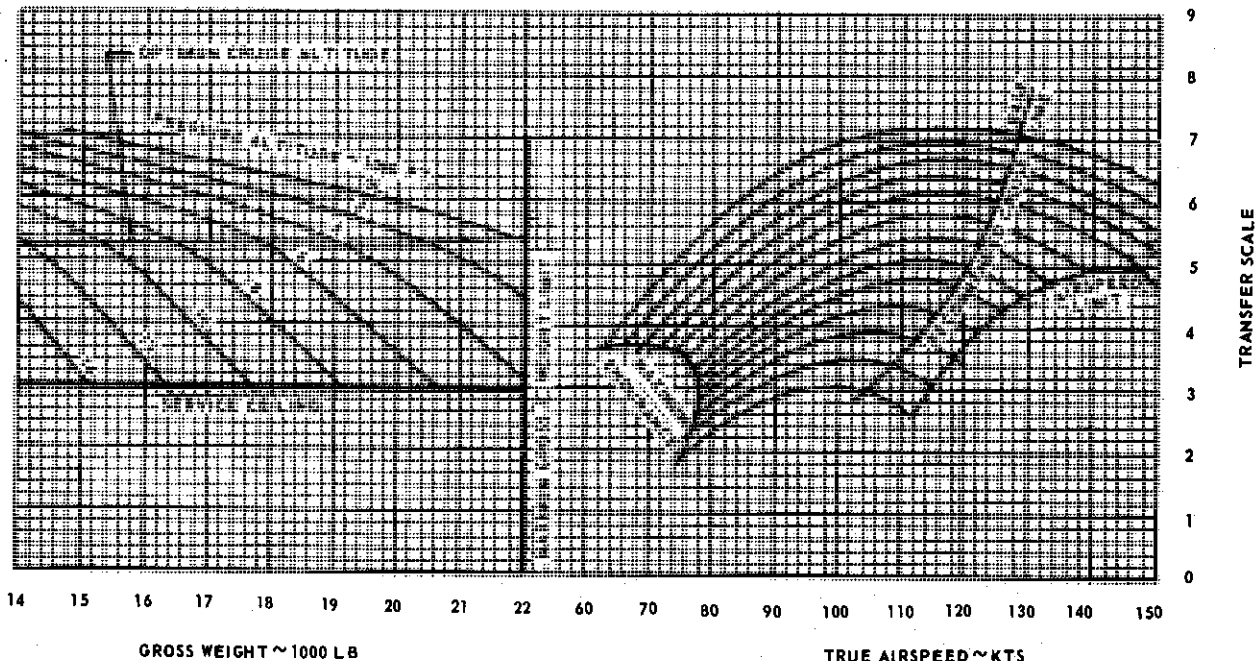
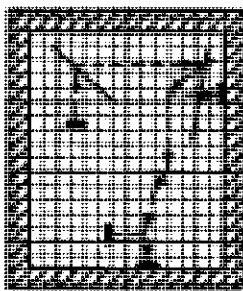
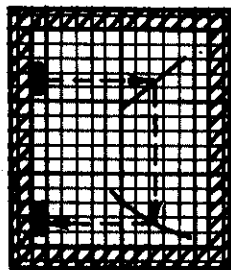


Figure A-28. Cruise (20°C to 40°C OAT) - Two Engines (Sheet 1 of 2)

CONDITIONS:
103% N_T
ZERO WIND
FOD SHIELD ON OR OFF



CRUISE

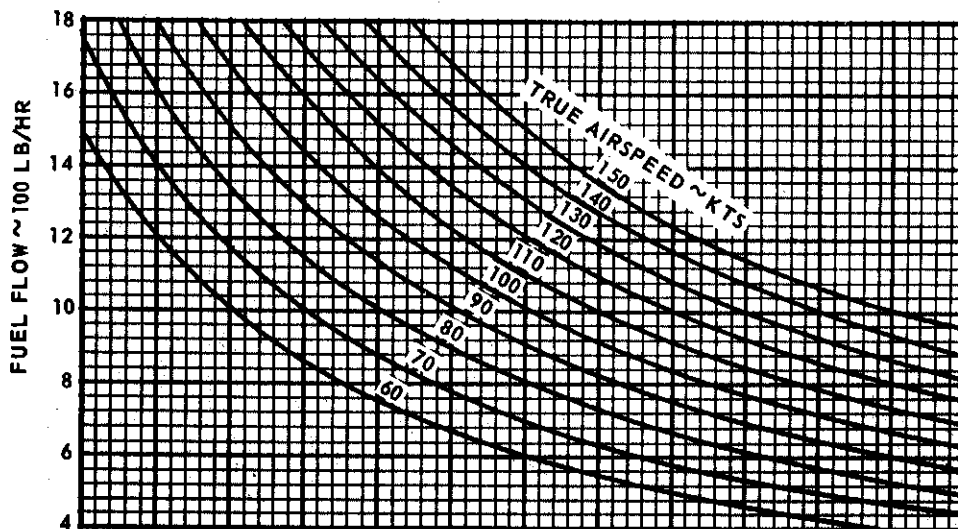
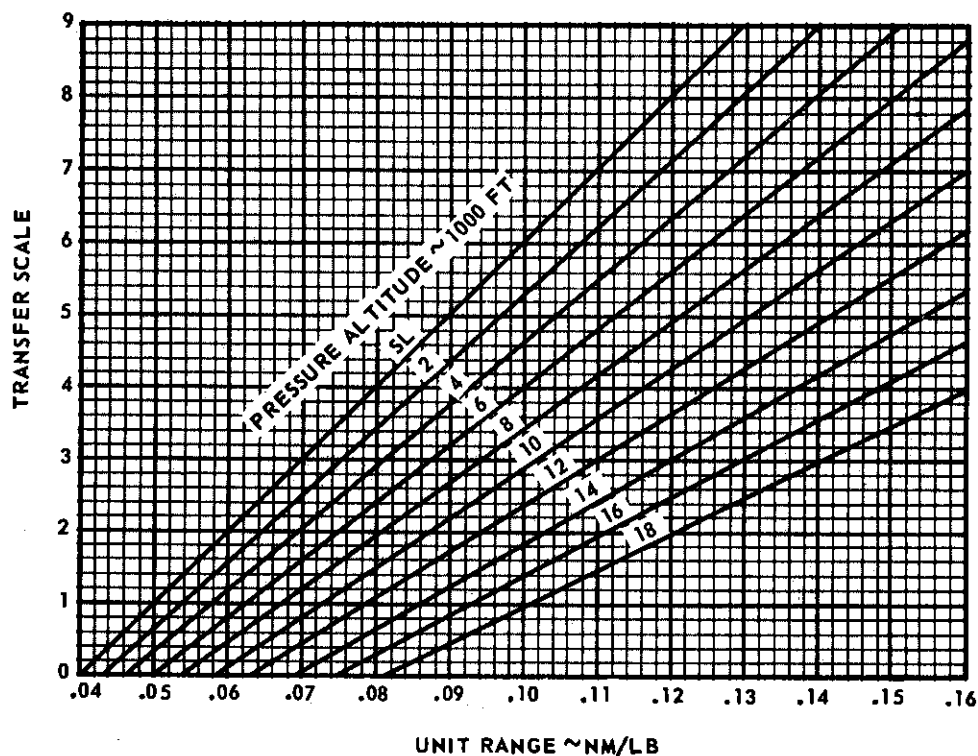
OAT BETWEEN 20°C AND 40°C
TWO ENGINES

MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17056.2 (C3)

Figure A-28. Cruise (20°C to 40°C OAT) - Two Engines (Sheet 2 of 2)

CRUISE

OAT BETWEEN 0°C AND 20°C
TWO ENGINES

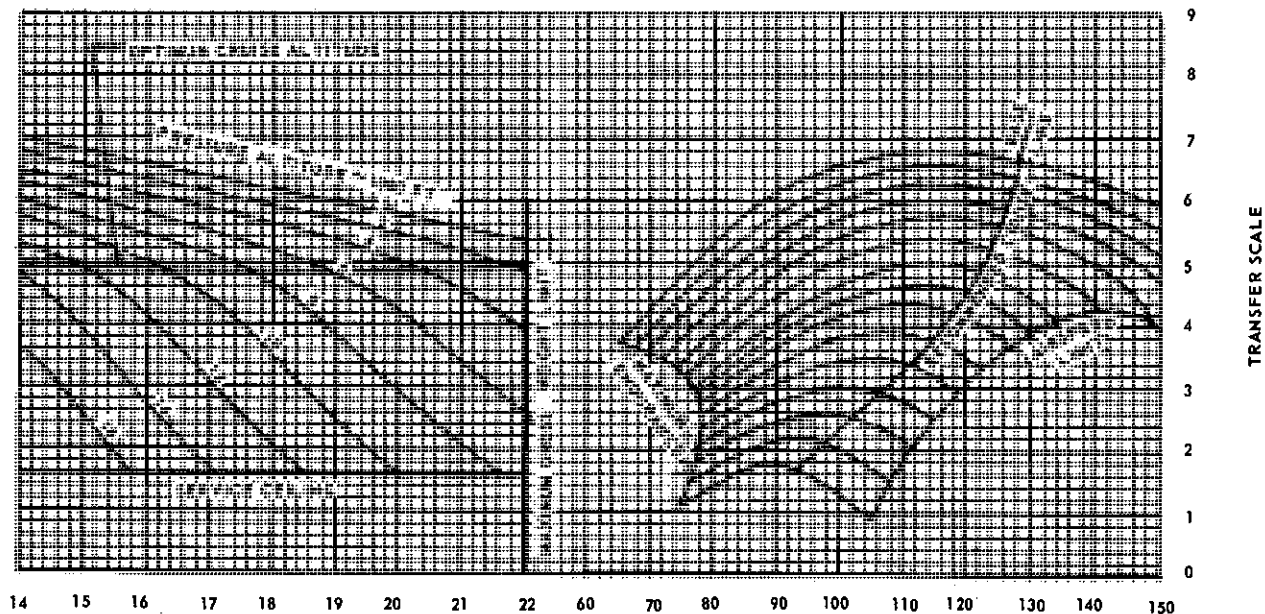
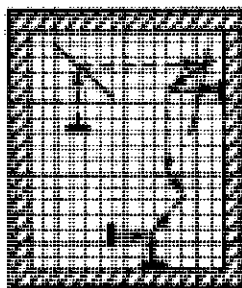
MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 2 NOVEMBER 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

CONDITIONS:
103% N_r
ZERO WIND
FOD SHIELD ON OR OFF



GROSS WEIGHT ~ 1000 LB

TRUE AIRSPEED ~ KTS

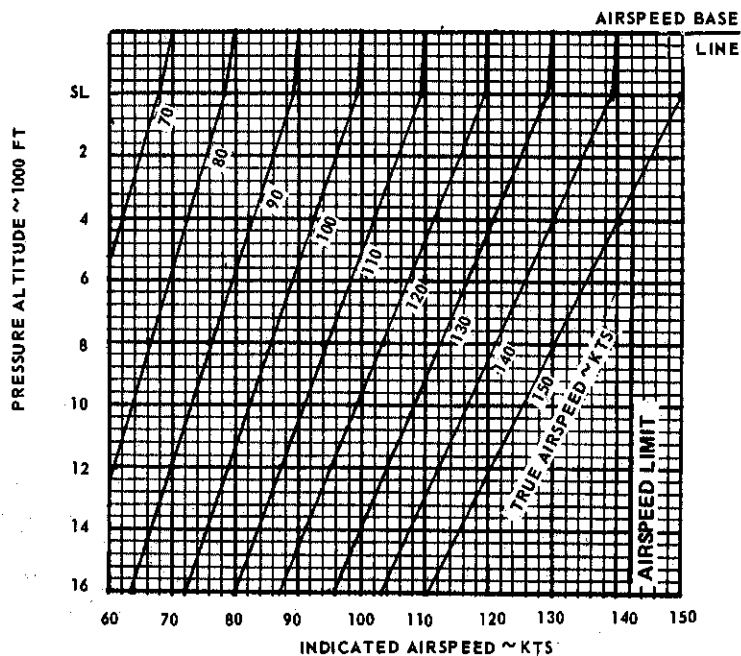
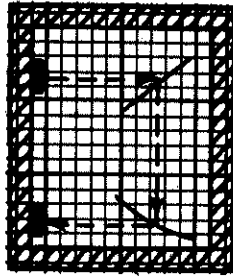


Figure A-29. Cruise (0°C to 20°C OAT) - Two Engines (Sheet 1 of 2)

CONDITIONS:103% N_1

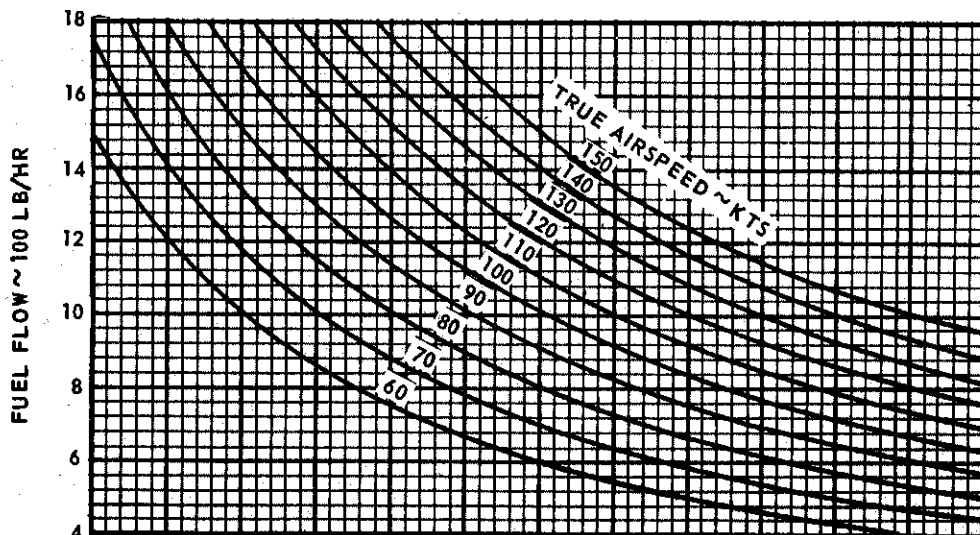
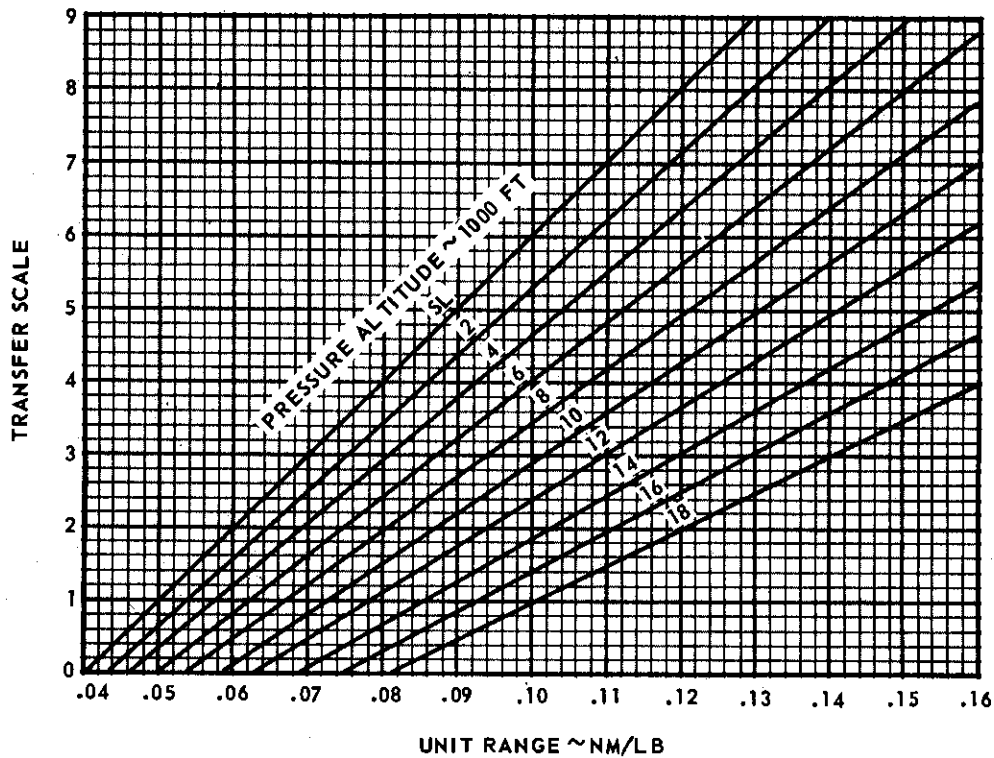
ZERO WIND

FOD SHIELD ON OR OFF

**CRUISE**OAT BETWEEN 0°C AND 20°C
TWO ENGINESMODEL
CH/HH-3EENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17057.2 (C3)

Figure A-29. Cruise (0°C to 20°C OAT) - Two Engines (Sheet 2 of 2)

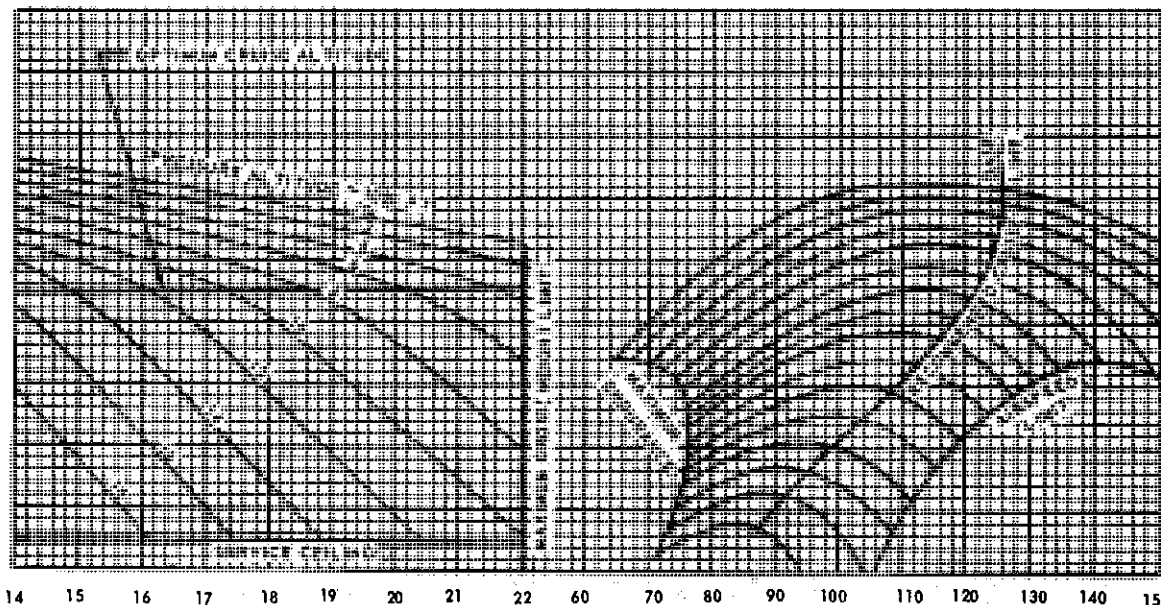
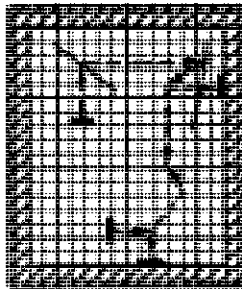
CRUISE

OAT BETWEEN -20°C AND 0°C
TWO ENGINES

MODEL CH/HH-3E ENGINE T58-GE-5

DATE: 2 NOVEMBER 1971
DATA BASIS: FLIGHT TEST (AIR FORCE)

CONDITIONS:
103% N_p
ZERO WIND
FOD SHIELD ON OR OFF



GROSS WEIGHT ~ 1000 LB

TRUE AIRSPEED ~ KTS

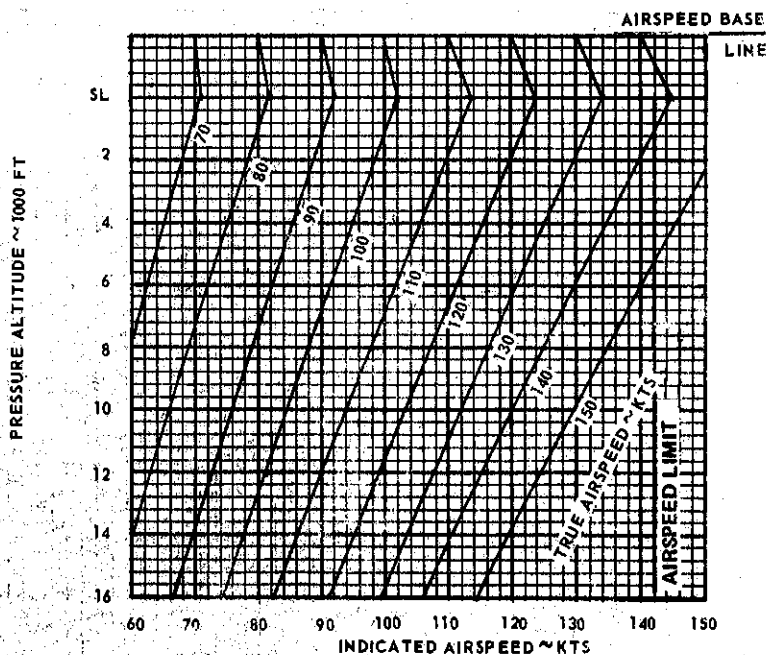
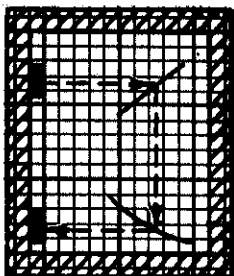


Figure A-30. Cruise (-20°C to 0°C OAT) - Two Engines (Sheet 1 of 2)

CONDITIONS:
103% N_F
ZERO WIND
FOD SHIELD ON OR OFF



CRUISE

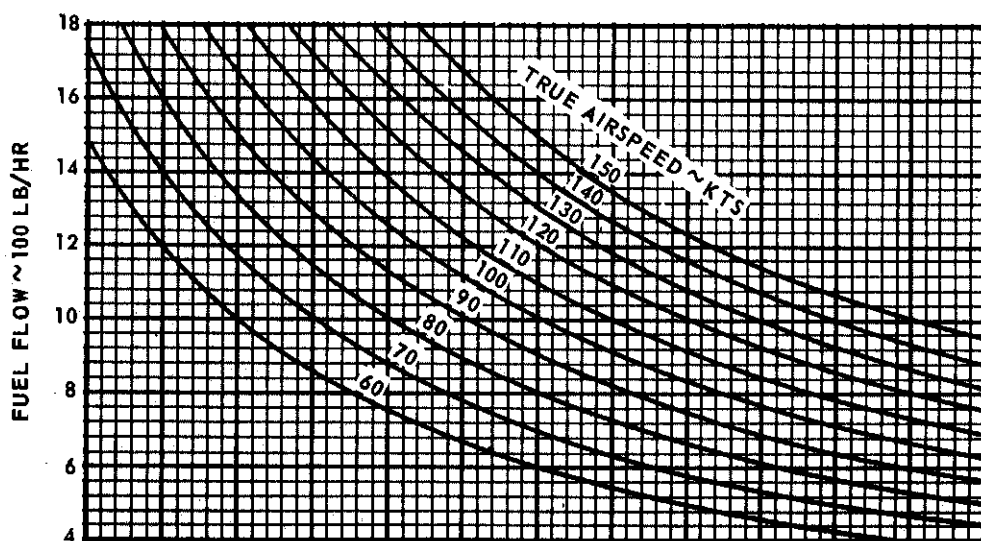
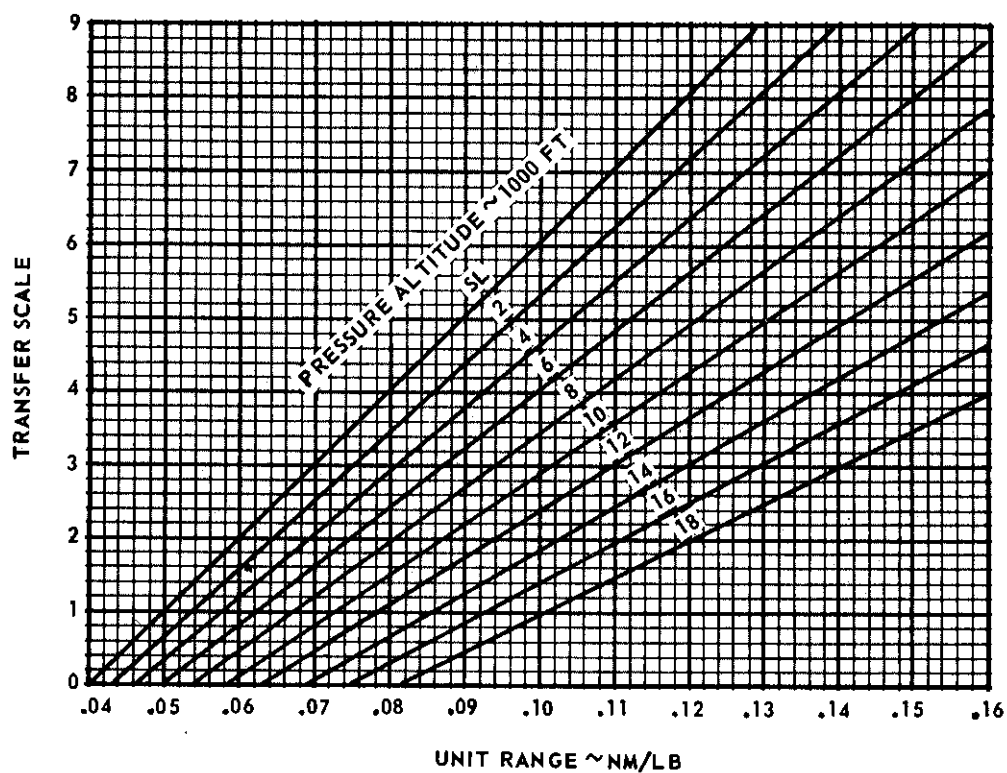
OAT BETWEEN -20°C AND 0°C
TWO ENGINES

MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17058.2 (C3)

Figure A-30. Cruise (-20°C to 0°C OAT) - Two Engines (Sheet 2 of 2)

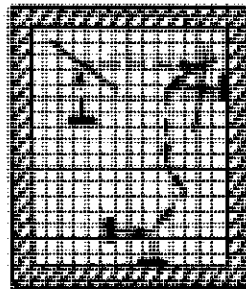
CRUISE

OAT BETWEEN -40°C AND -20°C
TWO ENGINES

MODEL
CH/HH-3E

ENGINE
T58-GE-5

CONDITIONS:
103% N₂
ZERO WIND
FOD SHIELD ON OR OFF



DATE: 2 NOVEMBER 1971
DATA BASIS: FLIGHT TEST (AIR FORCE)

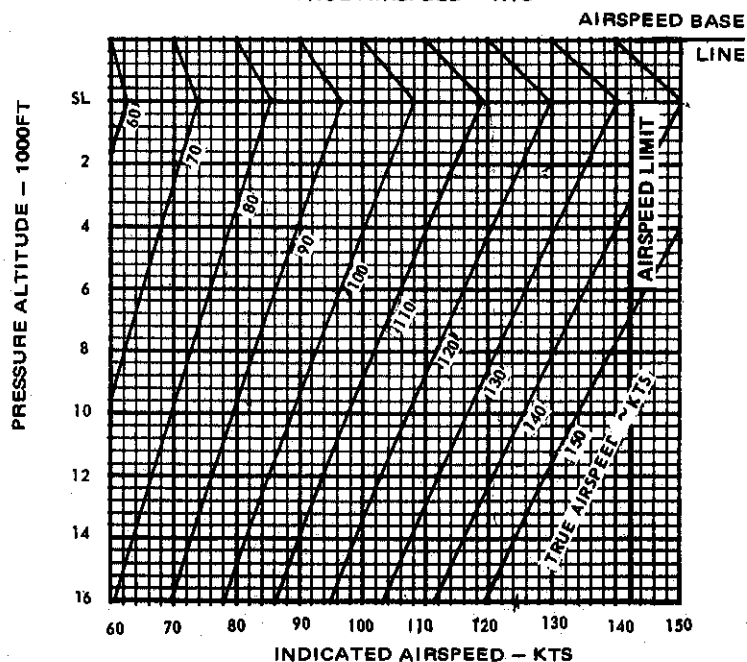
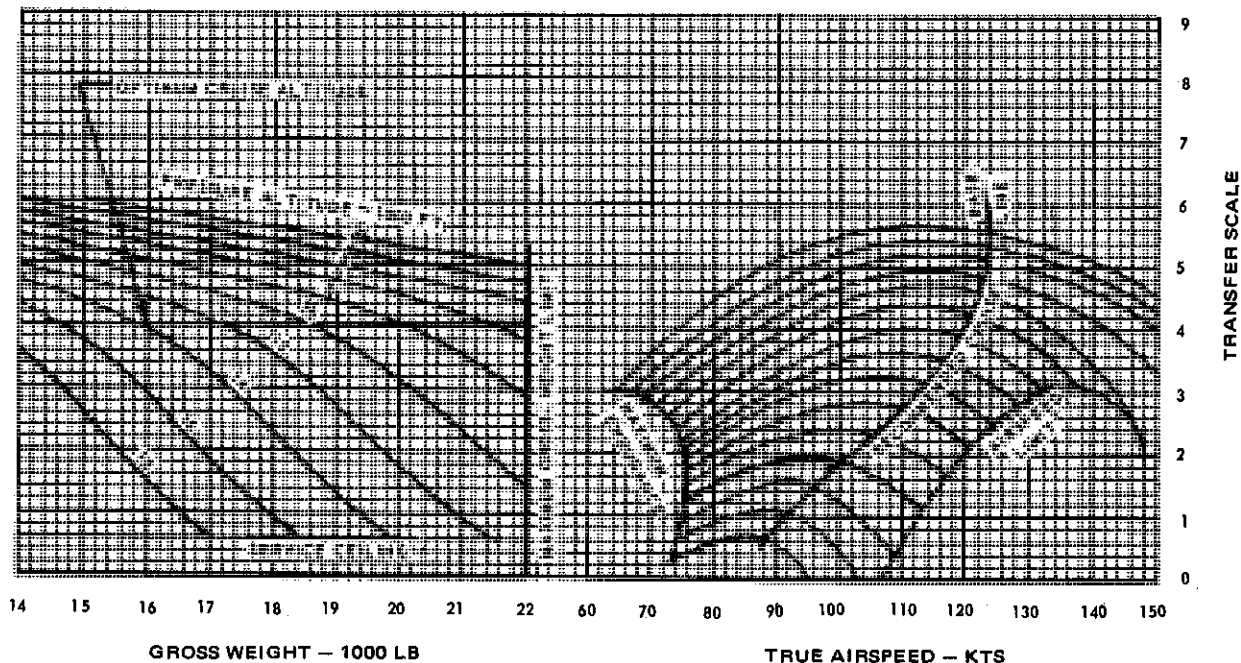
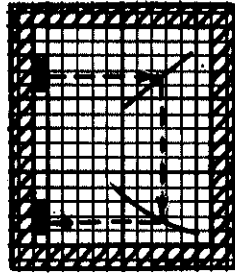


Figure A-31. Cruise (-40°C to -20°C OAT) - Two Engines (Sheet 1 of 2)

CONDITIONS:

103% N_T
 ZERO WIND
 FOD SHIELD ON OR OFF

**CRUISE**

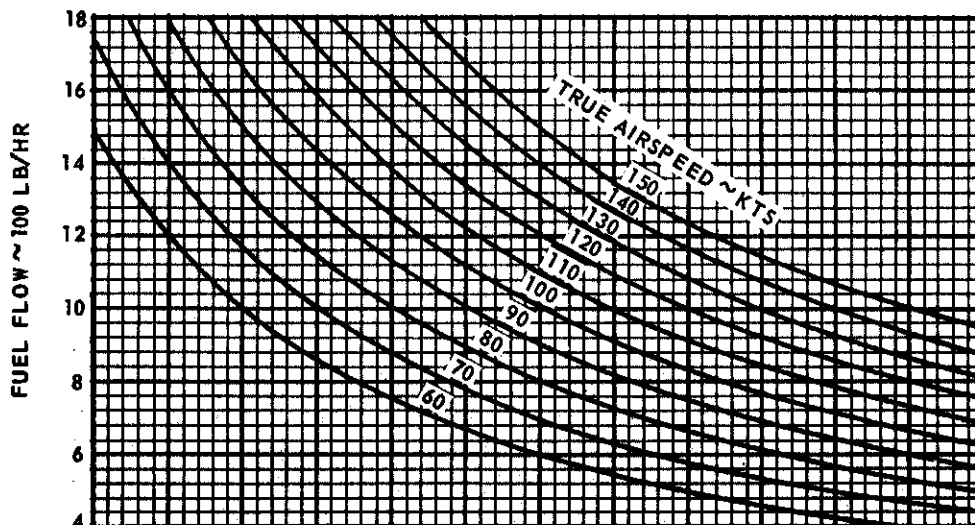
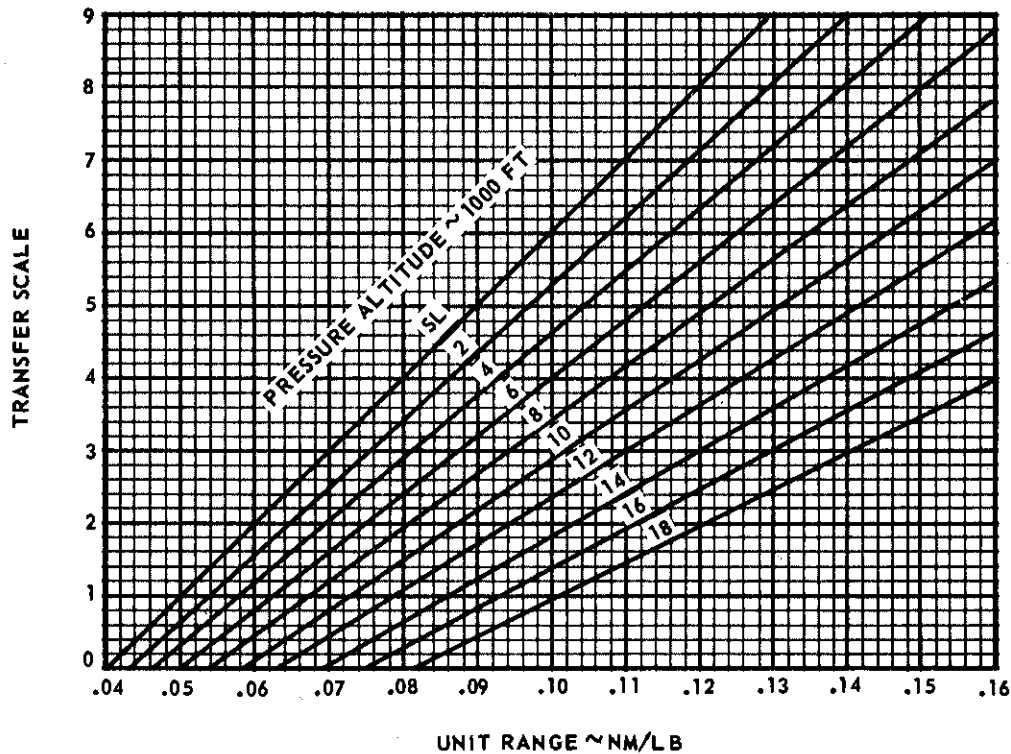
OAT BETWEEN -40°C AND -20°C
 TWO ENGINES

MODEL
 CH/HH-3E

ENGINE
 T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17059.2 (C3)

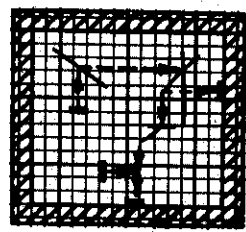
Figure A-31. Cruise (-40°C to -20°C OAT) - Two Engines (Sheet 2 of 2)

CRUISE

OAT BETWEEN 20°C AND 40°C
ONE ENGINE

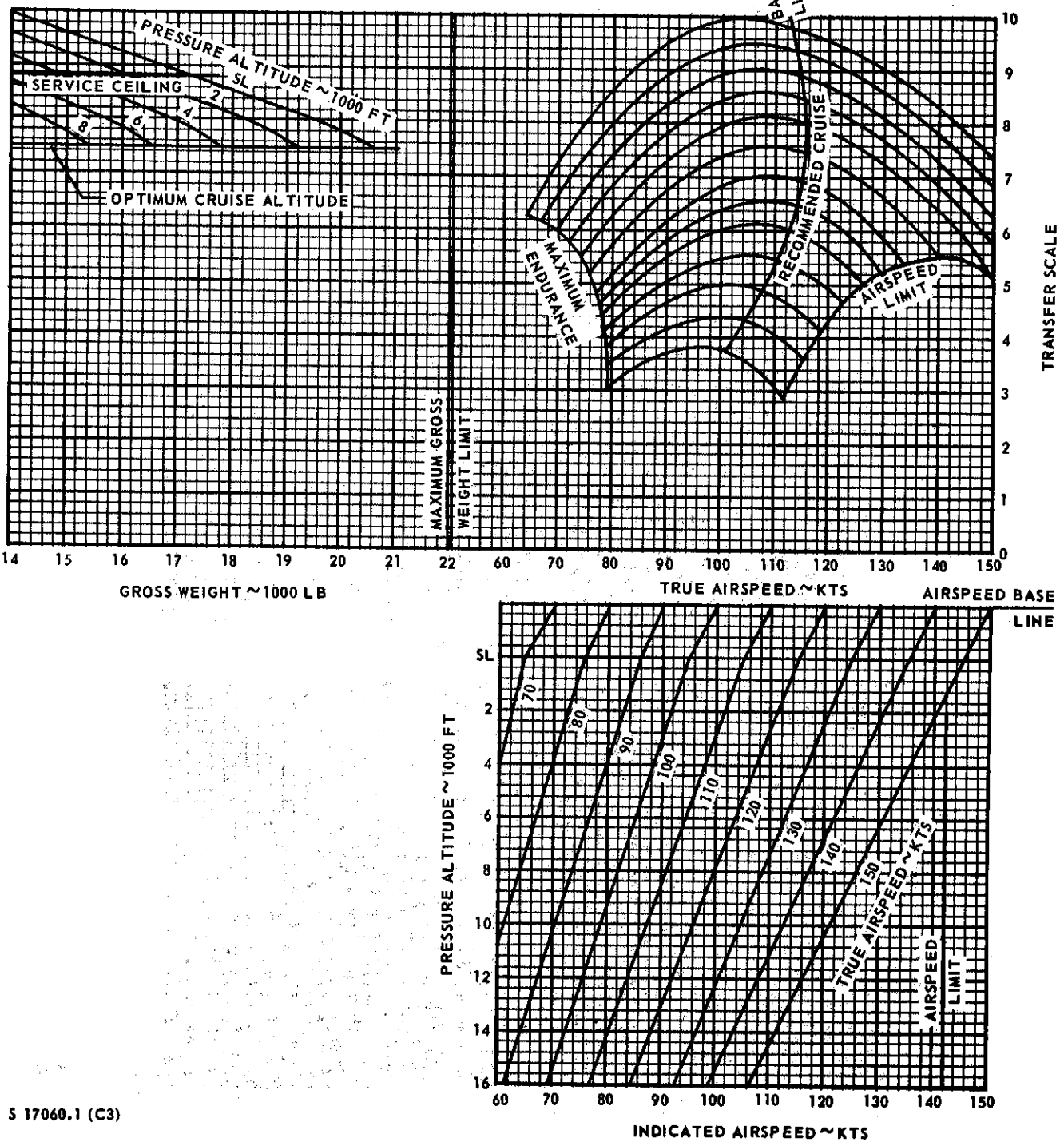
MODEL
CH/HH-3E

ENGINE
T58-GE-5



CONDITIONS:
103% N₁
ZERO WIND
FOD SHIELD ON OR OFF

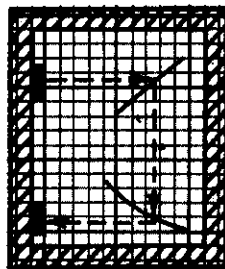
DATE: 15 APRIL 1971
DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17060.1 (C3)

Figure A-32. Cruise (20°C to 40°C OAT) - One Engine (Sheet 1 of 2)

CONDITIONS:
103% N_T
ZERO WIND
FOD SHIELD ON OR OFF



CRUISE

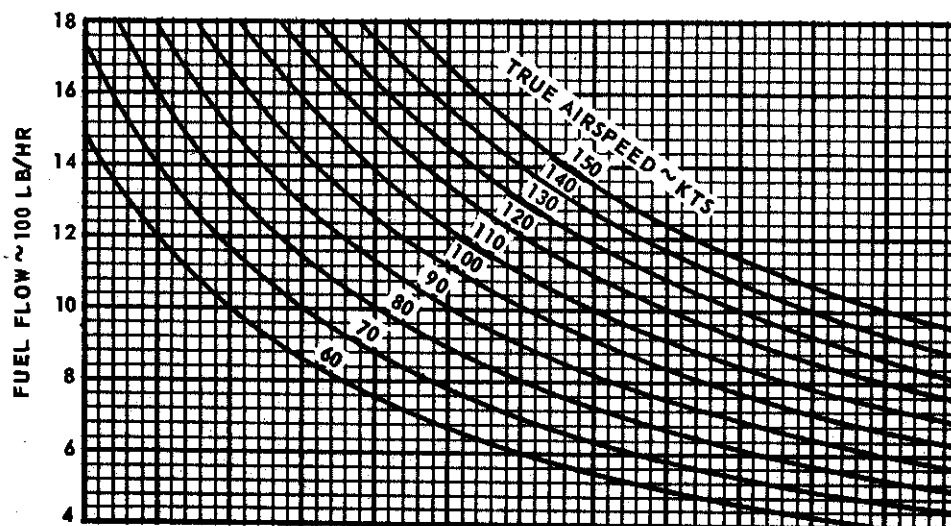
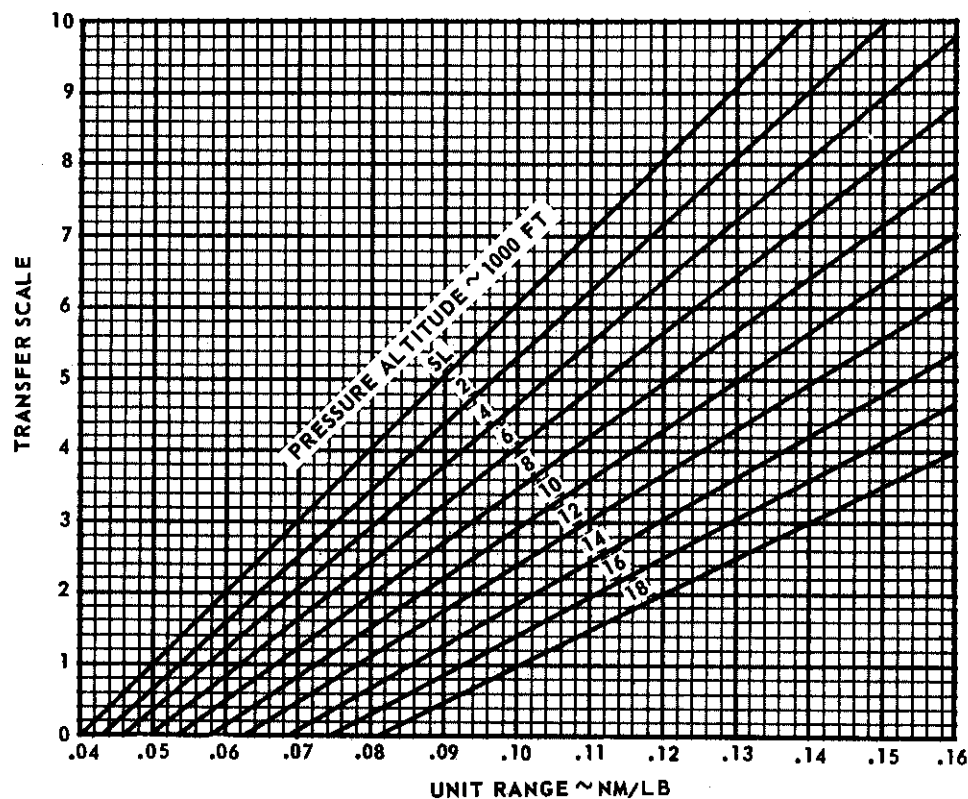
OAT BETWEEN 20°C AND 40°C
ONE ENGINE

MODEL
CH/HH-3E

ENGINE
T58-GE-5

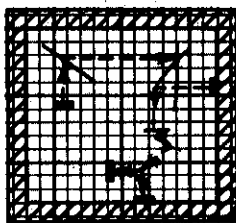
DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17060.2 (C3)

Figure A-32. Cruise (20°C to 40°C OAT) - One Engine (Sheet 2 of 2)

CRUISEOAT BETWEEN -40° AND 20°C
ONE ENGINEMODEL
CH/HH-3EENGINE
T58-GE-5

CONDITIONS:

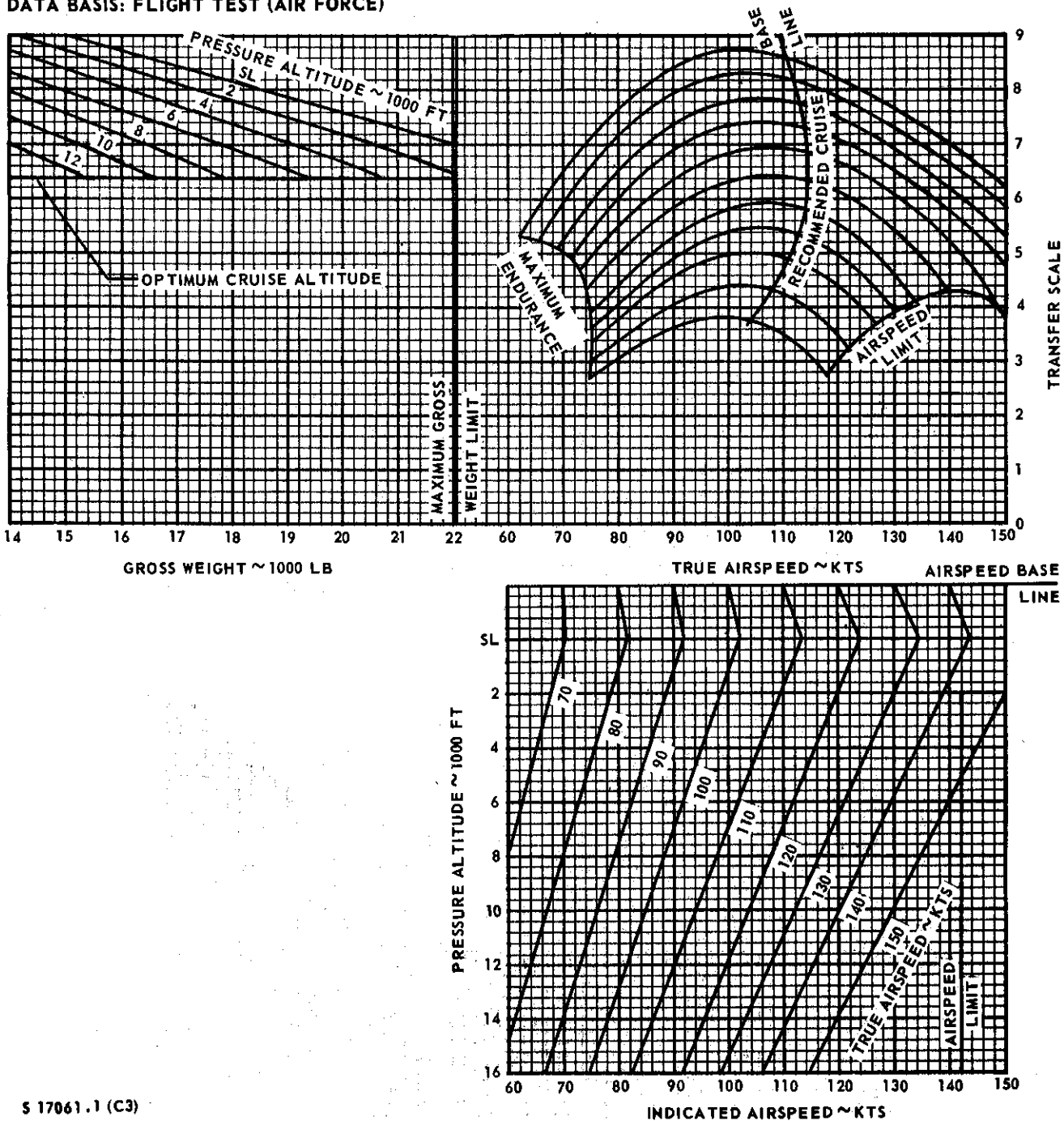
103% N_r

ZERO WIND

FOD SHIELD ON OR OFF

DATE: 15 APRIL 1971

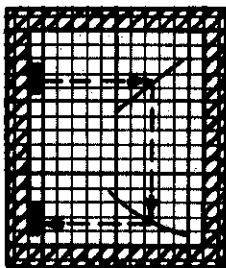
DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17061.1 (C3)

Figure A-33. Cruise (-40°C to 20°C OAT) - One Engine (Sheet 1 of 2)

CONDITIONS:
103% N_r
ZERO WIND
FOD SHIELD ON OR OFF



CRUISE

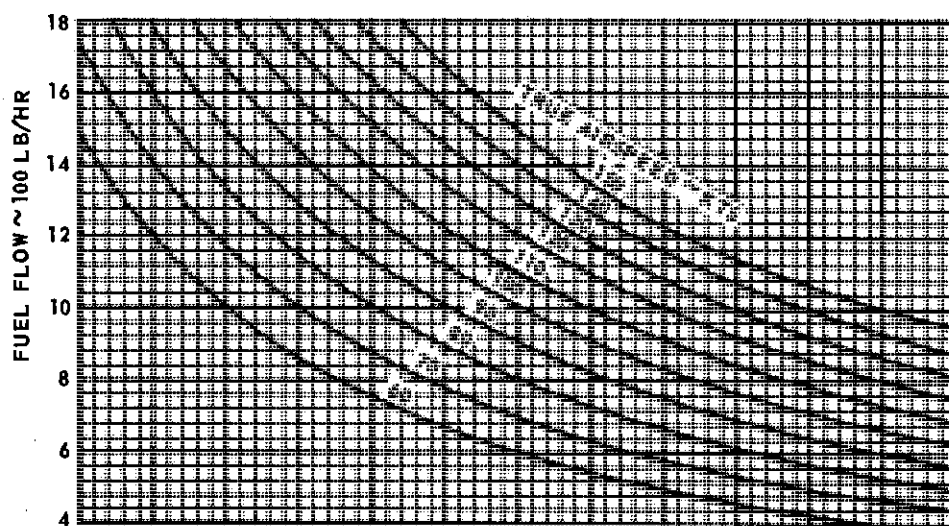
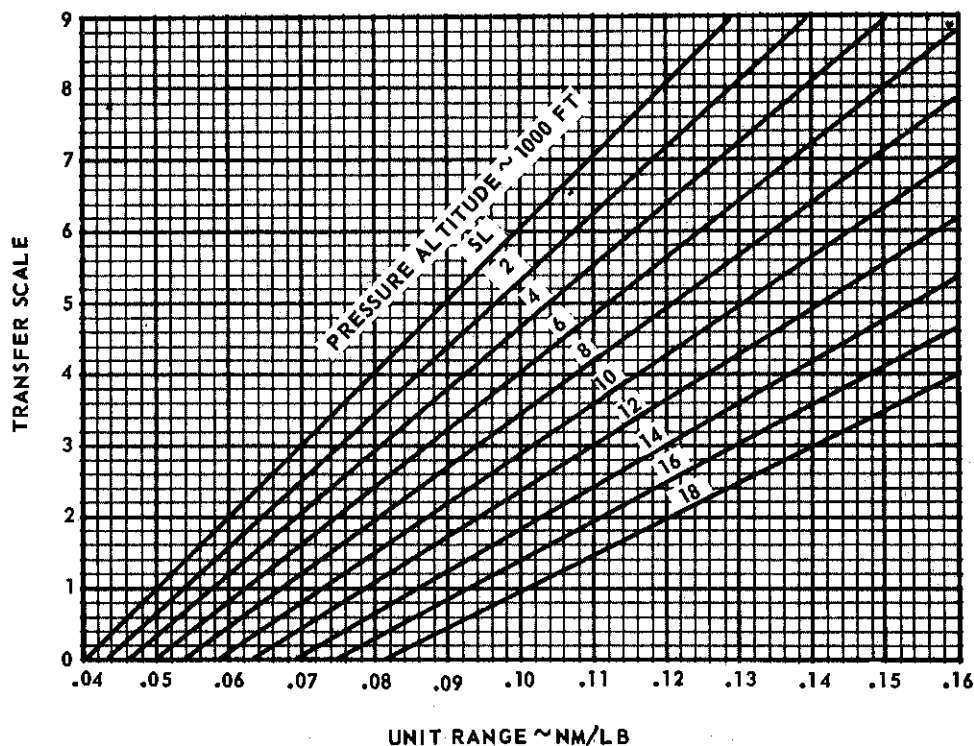
OAT BETWEEN -40°C AND 20°C
ONE ENGINE

MODEL
CH/HH-3E

ENGINE
T58-GE-5

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)



S 17061.2 (C3)

Figure A-33. Cruise (-40°C to 20°C OAT) - One Engine (Sheet 2 of 2)

SINGLE ENGINE CAPABILITY

ONE ENGINE

MODEL
CH/HH-3EENGINE
T58-GE-5

CONDITIONS:
70 KTS IAS
MILITARY POWER
103% Nr
FOD SHIELD
ON OR OFF
LANDING GEAR UP

DATE: 15 APRIL 1971

DATA BASIS: FLIGHT TEST (AIR FORCE)

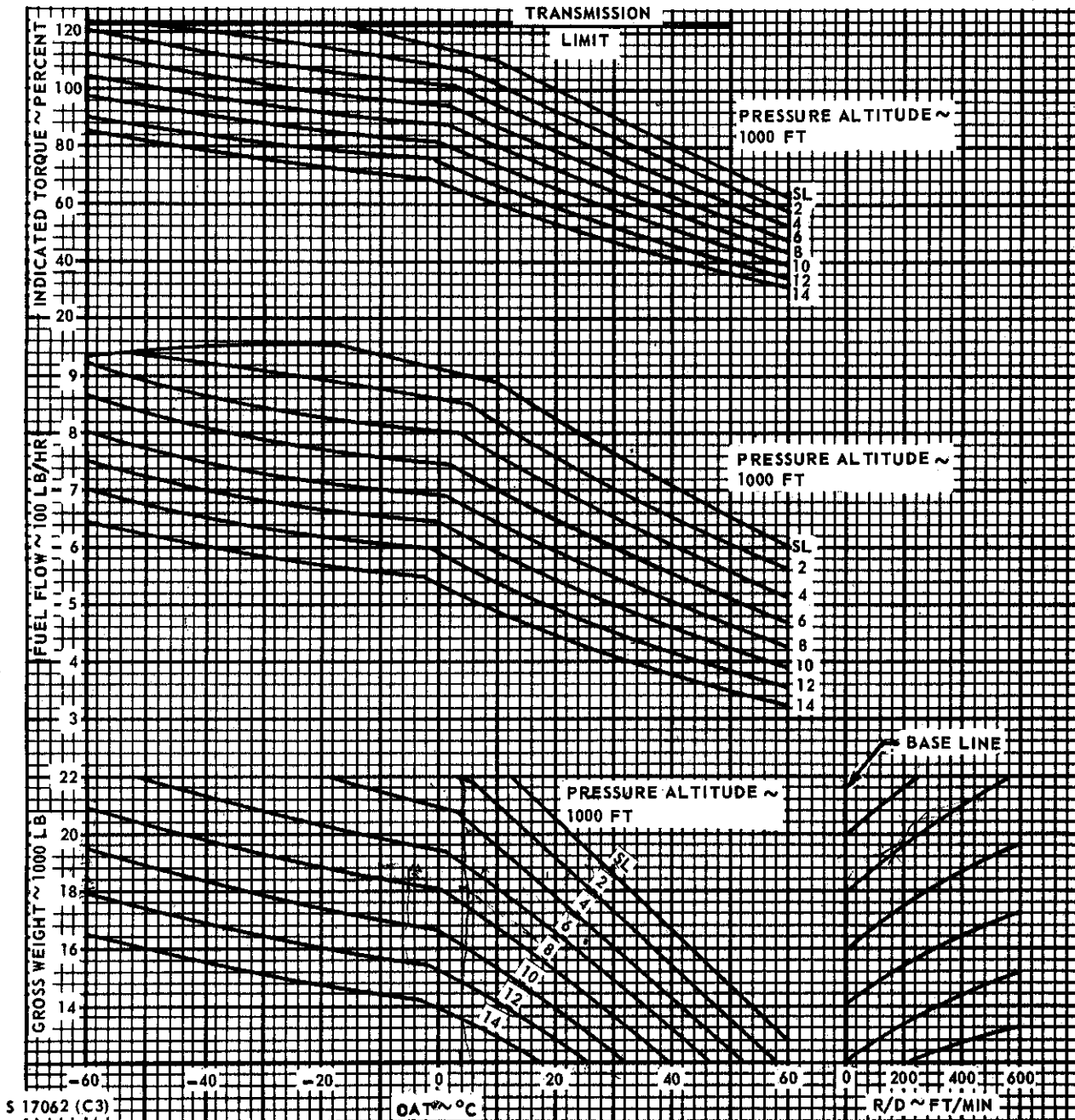
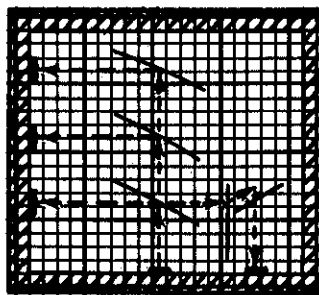
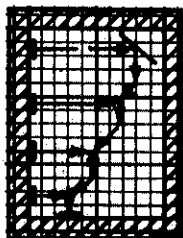


Figure A-34. Single Engine Capability - Military Power - One Engine

**BLADE STALL**

MODEL

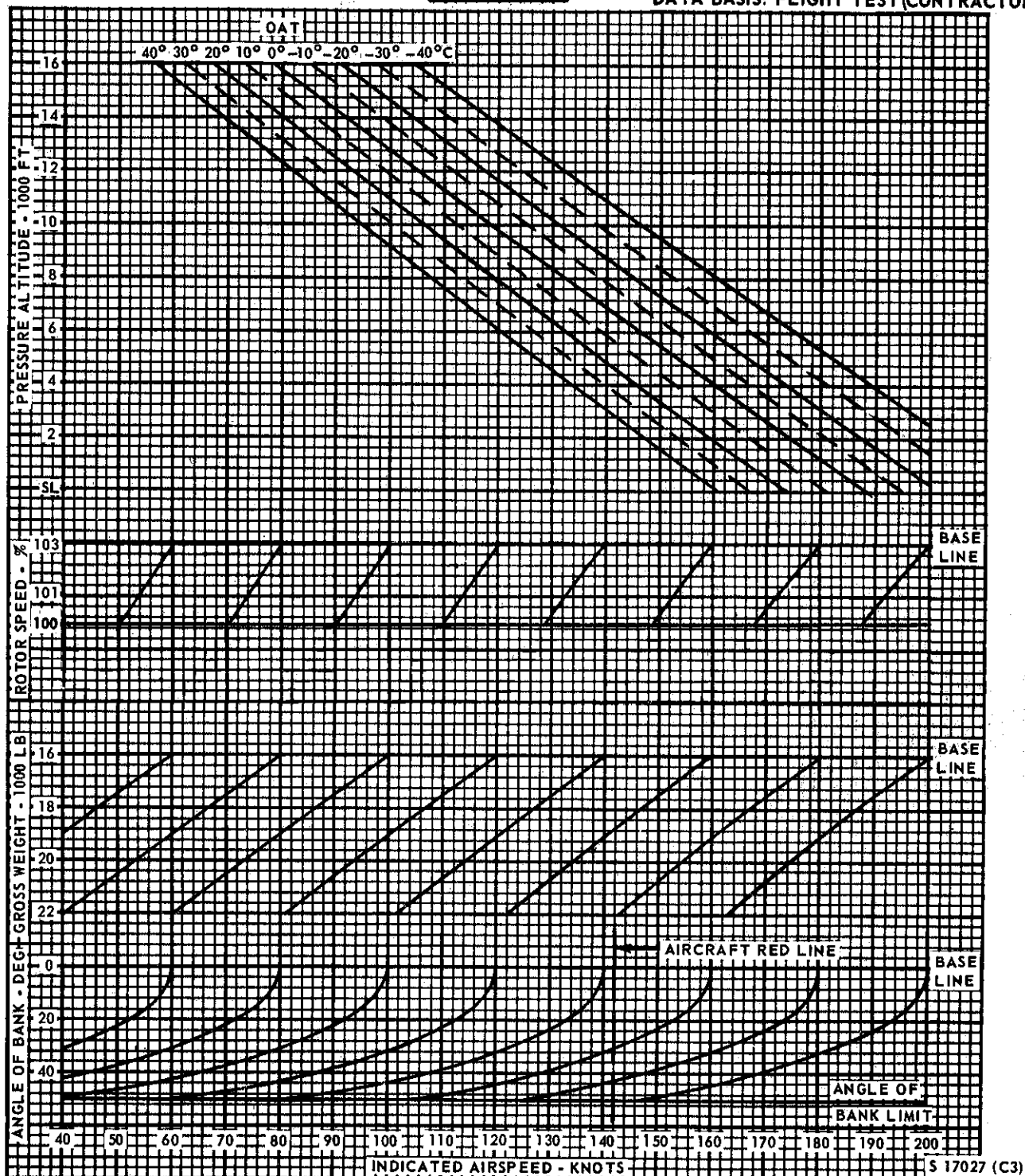
ENGINE

CH/HH-3E

T58-GE-5

DATE: 15 APRIL 1964

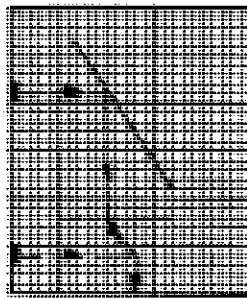
DATA BASIS: FLIGHT TEST (CONTRACTOR)



S 17027 (C3)

Figure A-35. Blade Stall

CONDITIONS:
UNACCELERATED
LEVEL FLIGHT
TIP MACH = 0.895



BLADE TIP MACH

MODELS
CH-3E & HH-3E

DATE: 27 JUNE 1972
DATA BASIS: FLIGHT TEST
(CONTRACTOR)

SEE 155-92

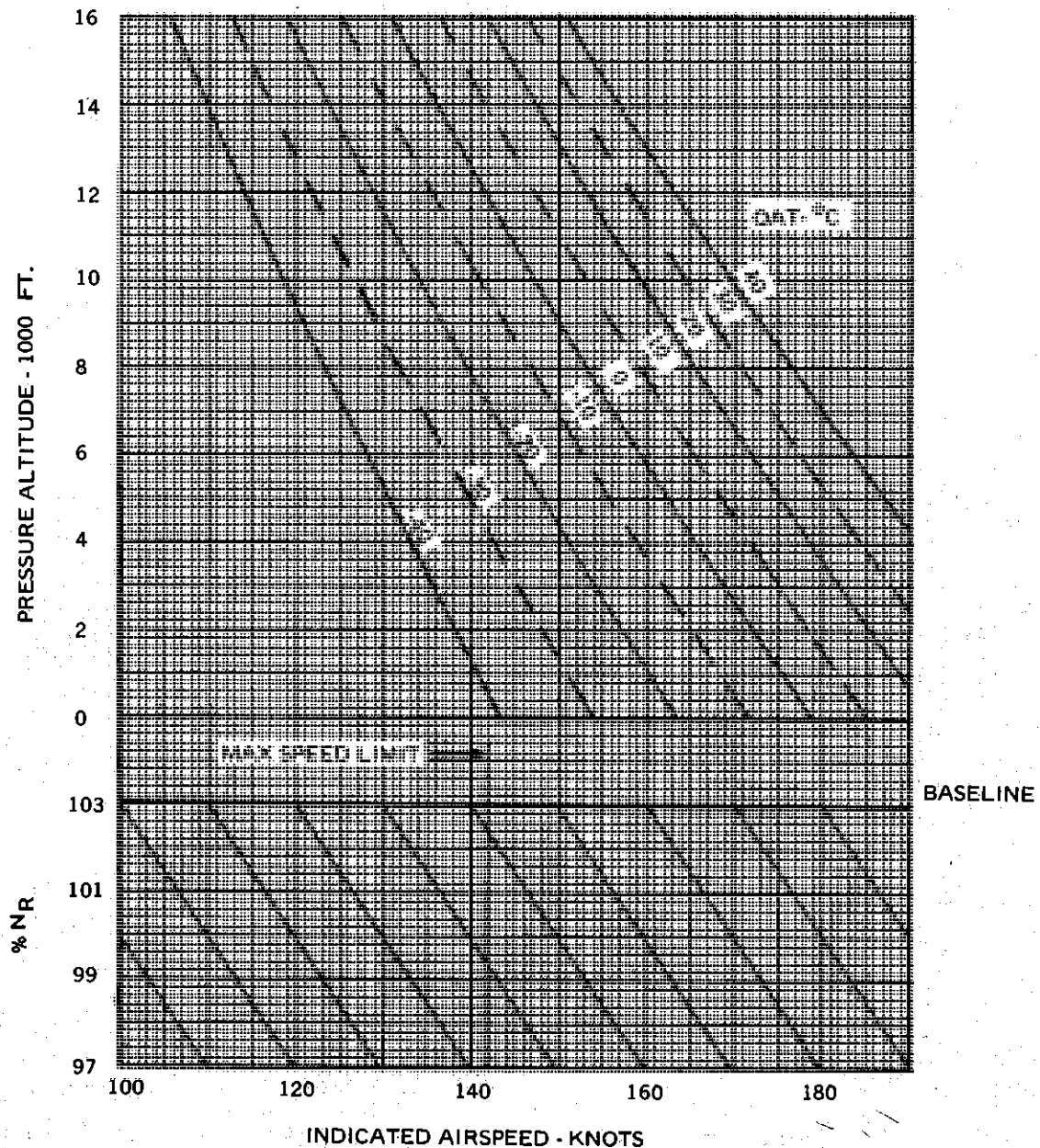
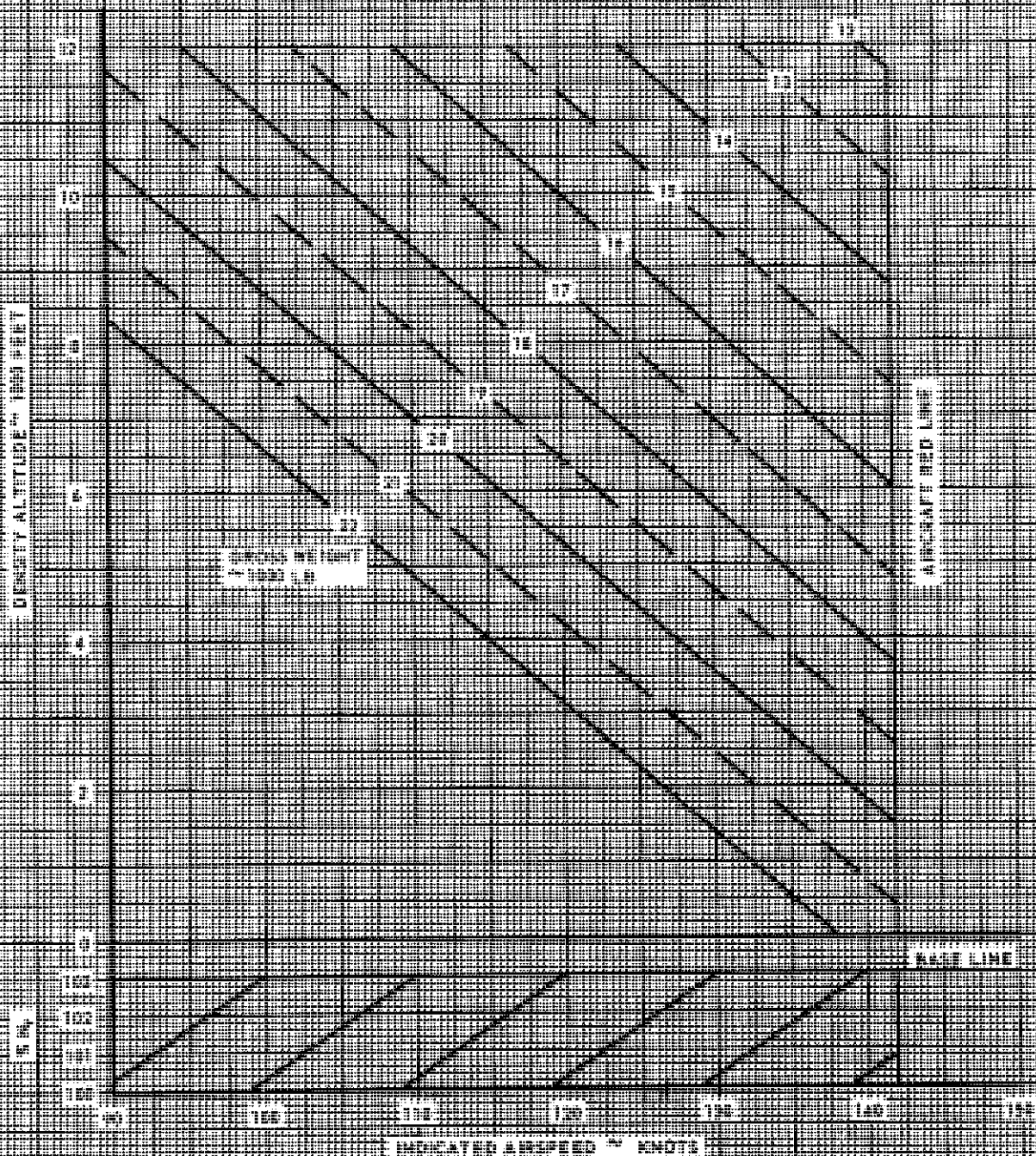


Figure A-36. Maximum Airspeed As Limited By Advancing Blade Tip Mach Number

MAXIMUM AIRSPEED CHART
VARIATION OF MAXIMUM SPEED (KNOTS, IAS) WITH DENSITY
ALTITUDE, GROSS WEIGHT AND ROTOR SPEED

MODEL: OH-35 & HH-35



S 16882 (RT)

Figure A-37. Maximum Airspeed Chart

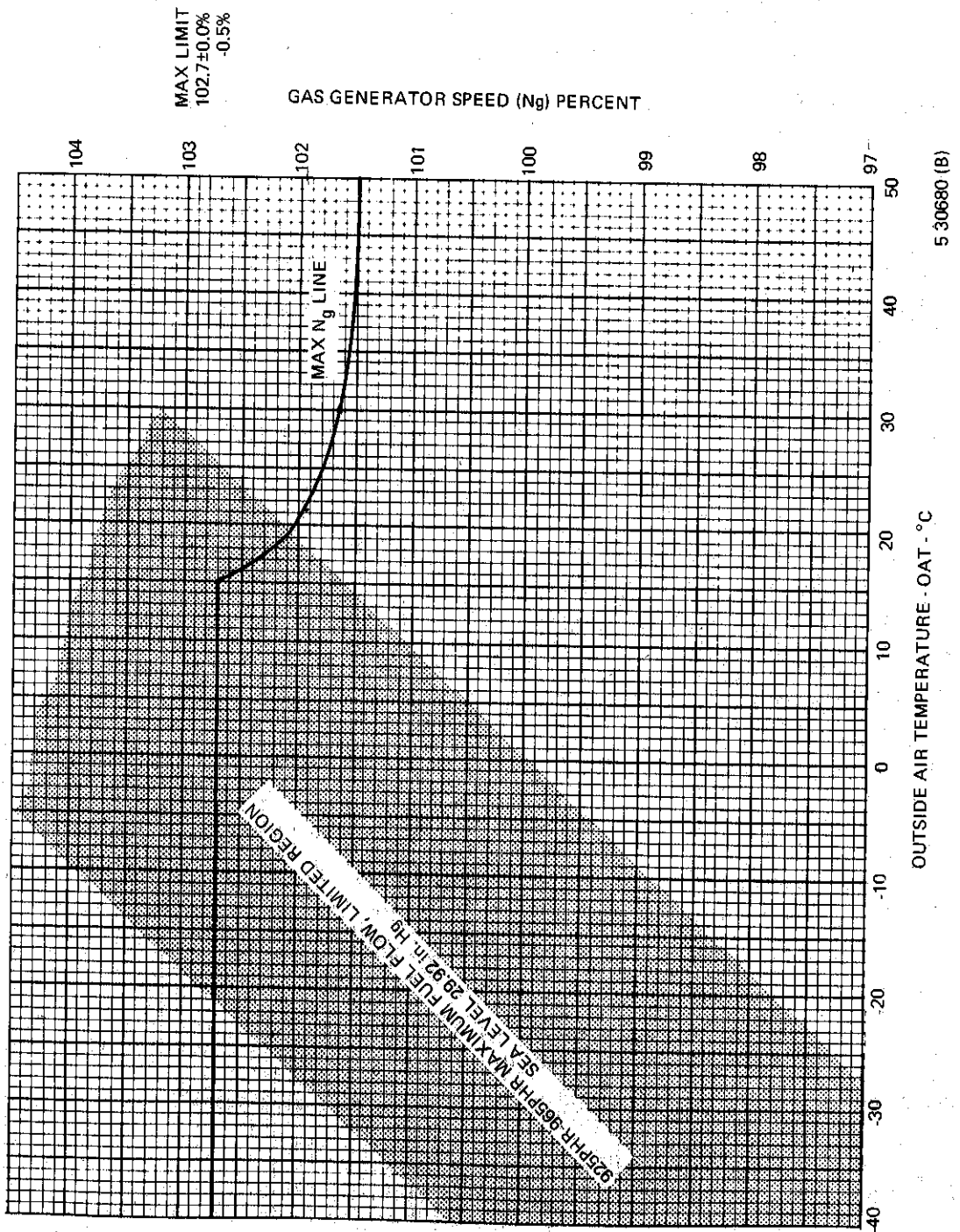


Figure A-38. Topping Chart (T58-GE-5)